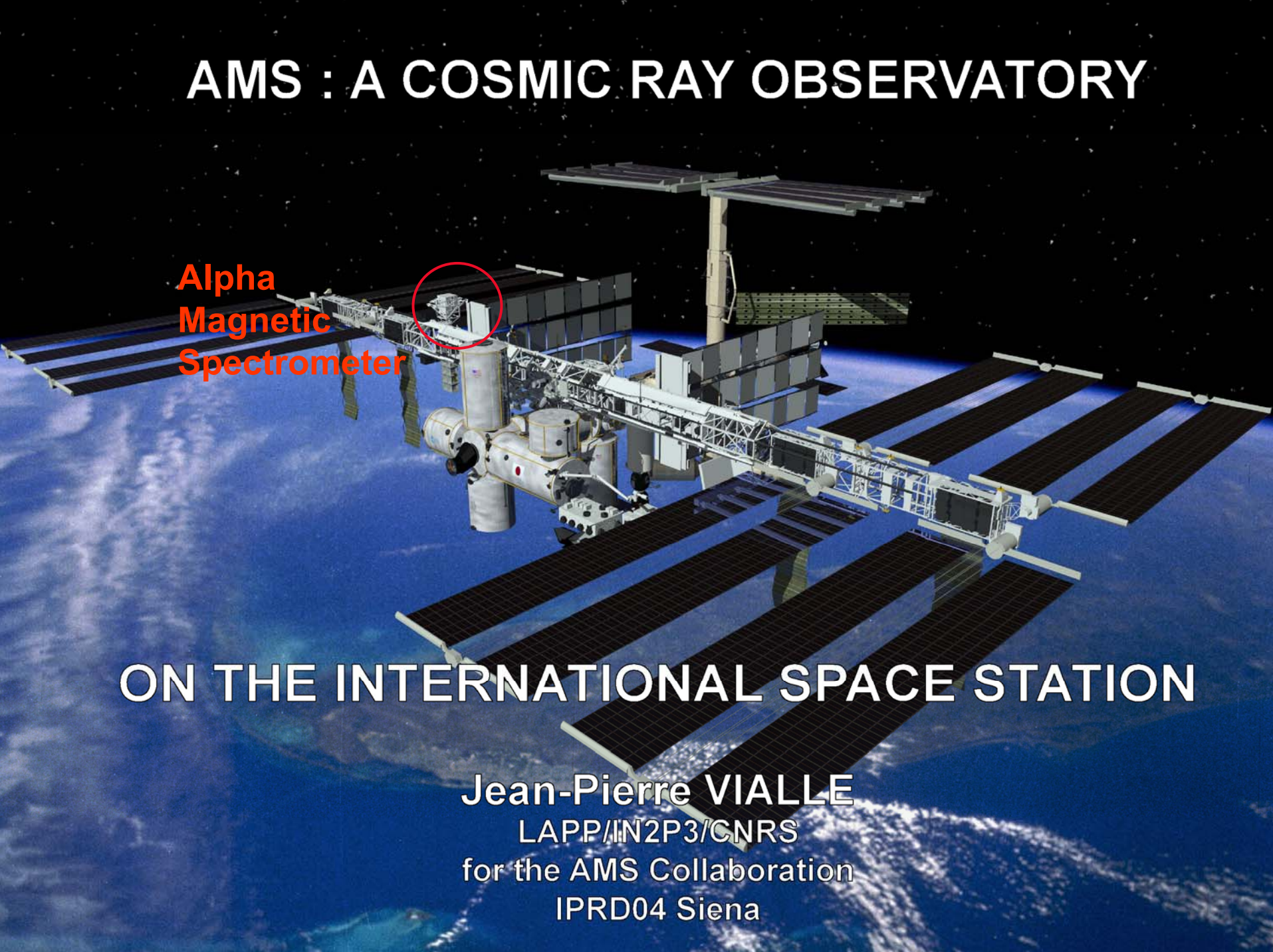


AMS : A COSMIC RAY OBSERVATORY

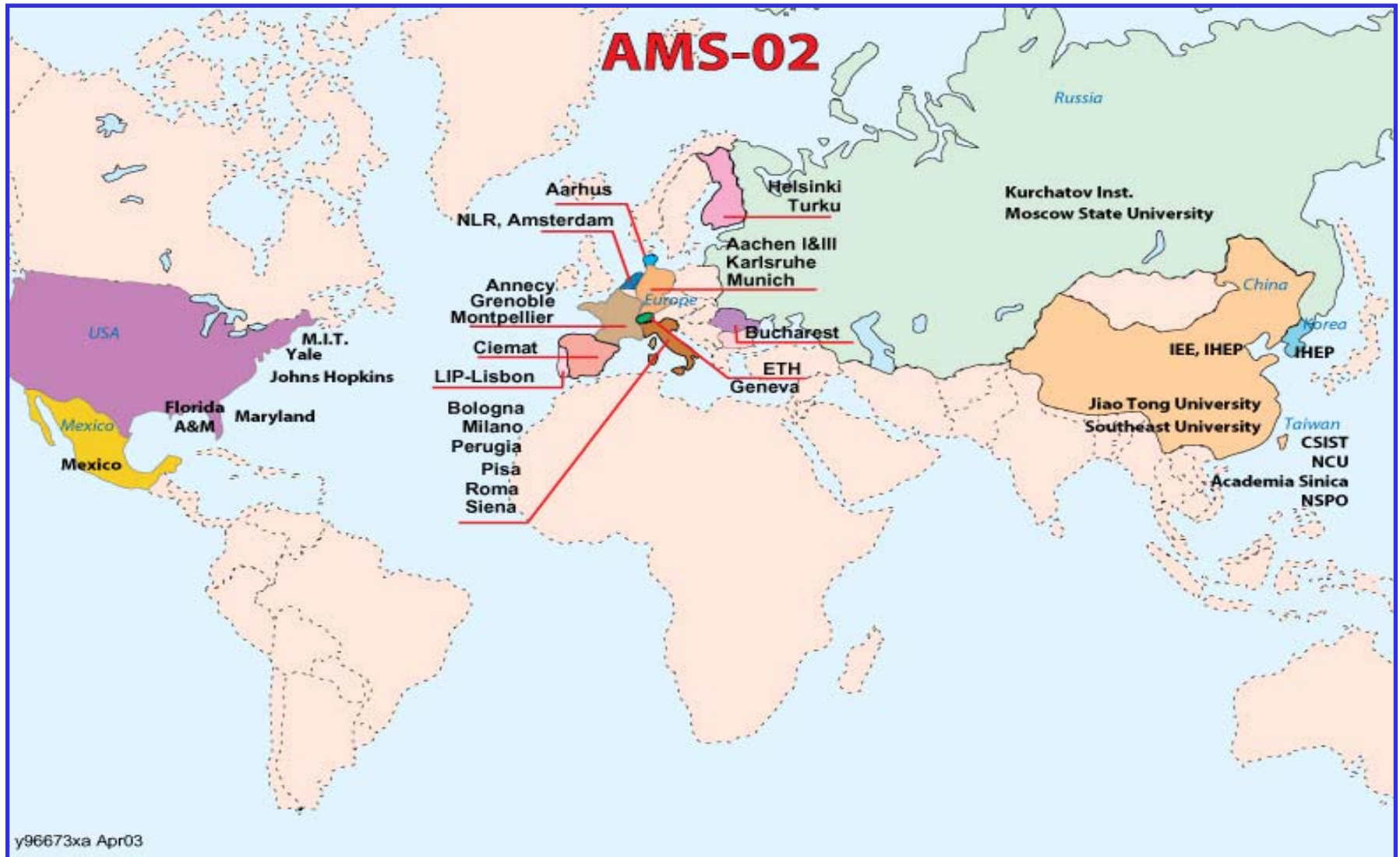
Alpha
Magnetic
Spectrometer



ON THE INTERNATIONAL SPACE STATION

Jean-Pierre VIALLE
LAPP/IN2P3/CNRS
for the AMS Collaboration
IPRD04 Siena

THE AMS-02 COLLABORATION: 15 COUNTRIES, 56 INSTITUTES



The Alpha Magnetic Spectrometer

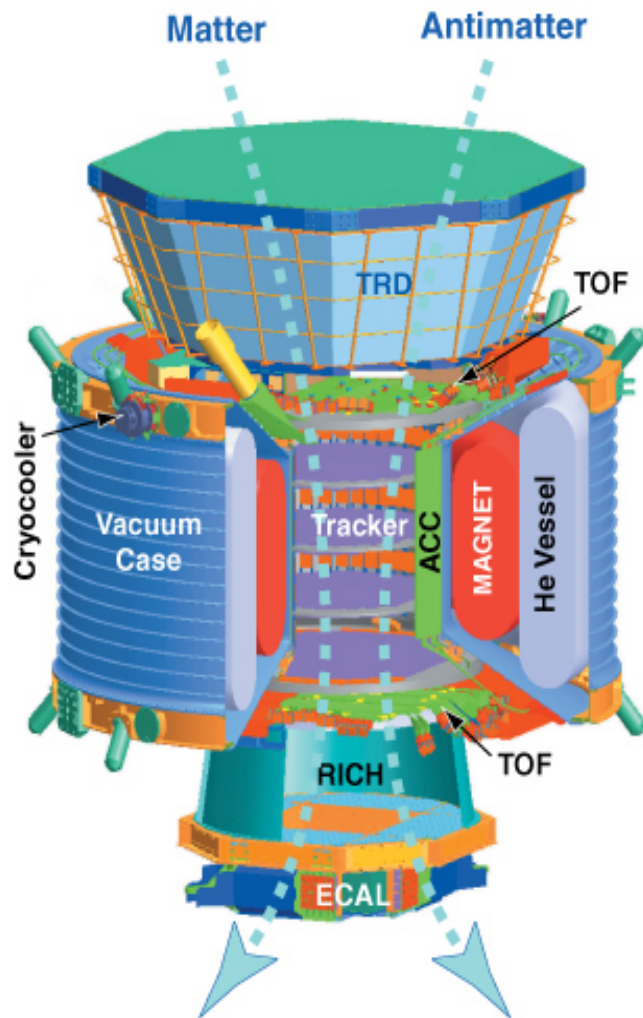
On the International Space Station as from
2007 for at least 3 years

PHYSICS GOALS

- Study of charged particles and nuclei in cosmic rays with high precision and high statistics in rigidity range 0.5 GV– few TV
- Direct search for antimatter (antihelium). Sensitivity 10^{-9} to antihelium/helium
- Indirect search for non-baryonic Dark Matter (neutralino $\chi + \chi \rightarrow e^+, \bar{p}, \gamma, +\dots$)
- High energy cosmic gamma-rays physics.

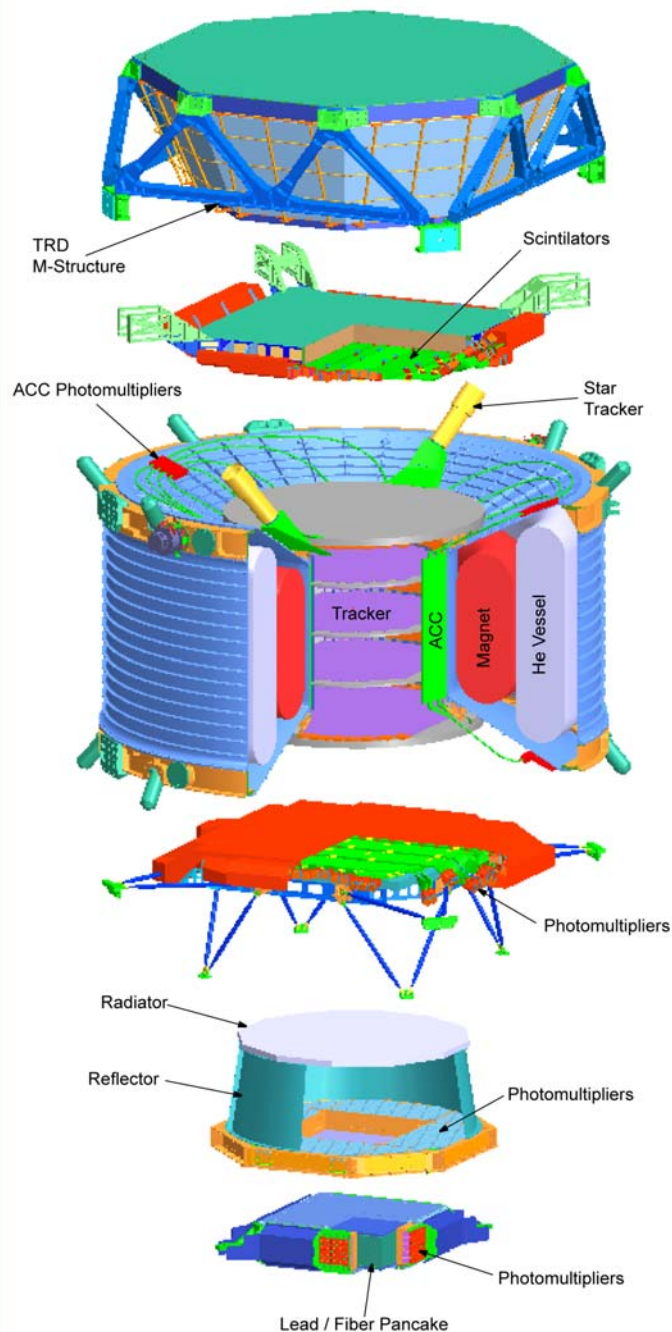
Total statistics expected above 10^{10} events.

AMS: A TeV Magnetic Spectrometer in Space (3m x 3m x 3m, 7t)



300,000 channels of electronics $\Delta t = 100 \text{ ps}$, $\Delta x = 10 \mu$

0.3 TeV	e^-	e^+	P	$\bar{\text{He}}$	γ
TRD					
TOF					
Tracker					
RICH					
Calorimeter					



TRD:
Transition
Radiation
Detector

TOF: (s1,s2)
Time of Flight
Detector

MG:
Magnet

TR:
Silicon Tracker

ACC:
Anticoincidence
Counter

AST:
Amiga Star
Tracker

TOF: (s1,s2)
Time of Flight
Detector

RICH:
Ring Image
Cherenkov Counter

EMC;
Electromagnetic
Calorimeter

- Transition Radiation Detector. 20 planes
- Time of Flight. 4 planes of scintillating counters
- Double-sided Si strip tracker planes inside superconducting magnet. 8 Layers
- Rich: Ring Imaging Cherenkov detector
- 3D Electromagnetic calorimeter

MAIN DESIGN CHARACTERISTICS

- Minimum amount of matter (X_0) before ECAL
- Acceptance $0.5 \text{ m}^2 \cdot \text{Sr}$ -> anti-He search
- Velocity measurement $\Delta\beta/\beta = 0.1 \%$ to distinguish ^9Be , ^{10}Be , ^3He , ^4He isotopes.
- Rigidity $R = pc/|Z|e$ (GV) proton resolution 20% at 0.5 TV and Helium resolution of 20% at 1 TV.
- Antihelium/Helium identification factor 10^{10} .

Multiple and independent measurements to reach performances required :

- $|Z|$ measured from Tracker, RICH, TOF.
- Sign of charge Z measured from tracker (8 points).
- Velocity β measured from TOF, RICH.
- Hadron/electron separation from TRD, ECAL.

Constraints on design due to launch and space

- ❑ **Vibration (6.8 G rms) and G-Forces (17G)**
- ❑ **Thermal Environment (day/night: $\Delta T \sim 100^\circ\text{C}$)**
- ❑ **Vacuum: $< 10^{-10}$ Torr**
- ❑ **Radiation: Ionizing Flux $\sim 1000 \text{ cm}^{-2}\text{s}^{-1}$**
- ❑ **Orbital Debris and Micrometeorites**

- ❑ **Limitation : Weight (14 809 lb) and Power (2000 W)**
- ❑ **Reliable for more than 3 years – Redundancy**
- ❑ **Must operate without services and human Intervention**

ALPHA MAGNETIC SPECTROMETER AMS01 MISSION

STS91 flight on Discovery 2-12 June 1998

10 days in space. 10^8 cosmic ray triggers

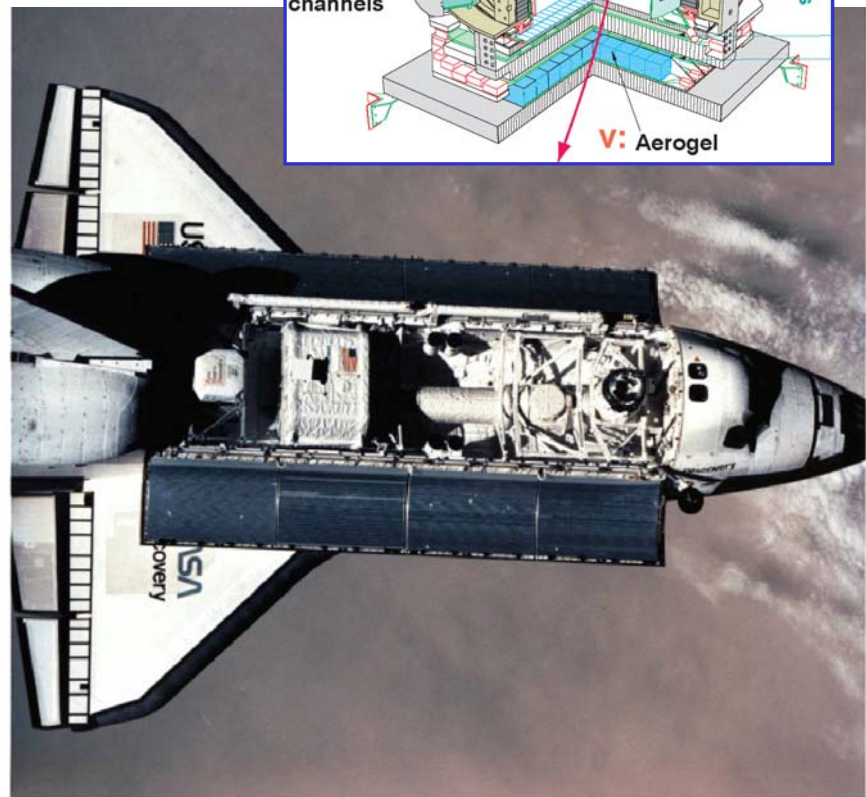
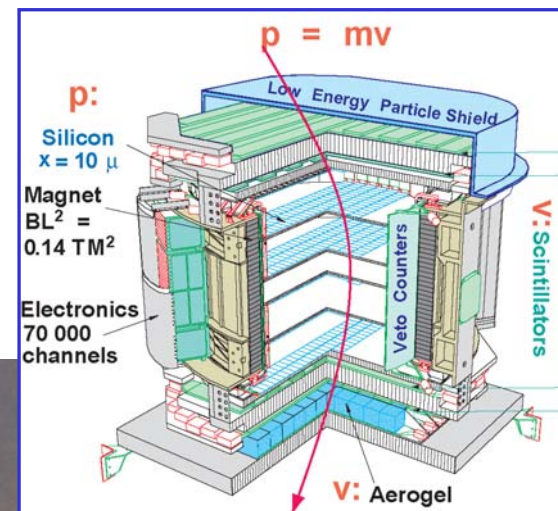
Several important physics results published
(Overview in Physics Reports 366 (2002))

Limit on primordial antimatter :
Search for antihelium in C.R. PL B461 (1999)

Measurement of primary fluxes p, He, e-, e+, ...,
detection of secondary fluxes, geomagnetic field
effect and particles trapping:

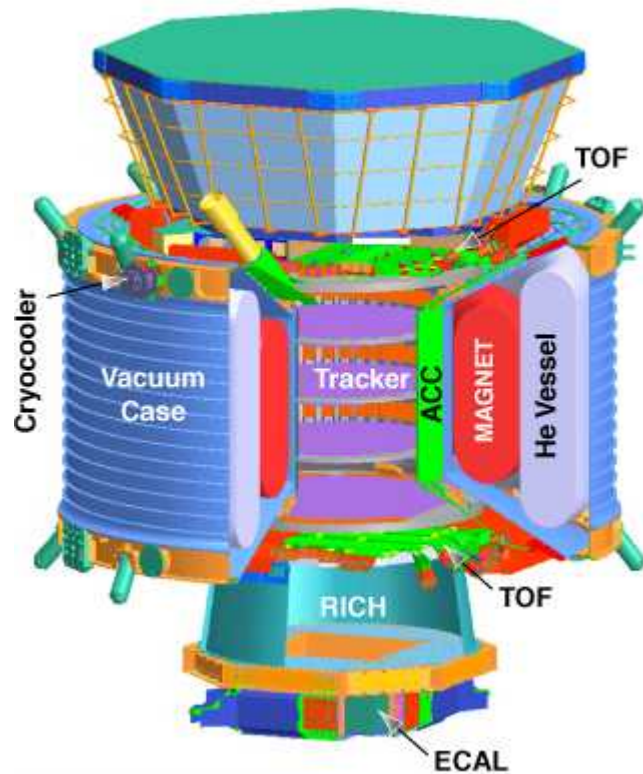
Protons in near earth orbit PL B472 (2000)
Leptons in near earth orbit PL B484 (2000)
Cosmic protons PL B490 (2000)
Helium in near earth orbit PL B494 (2000)

Use of results for other fluxes calculations:
A 3D simulation of atmospheric neutrinos
PRD68 (2003)



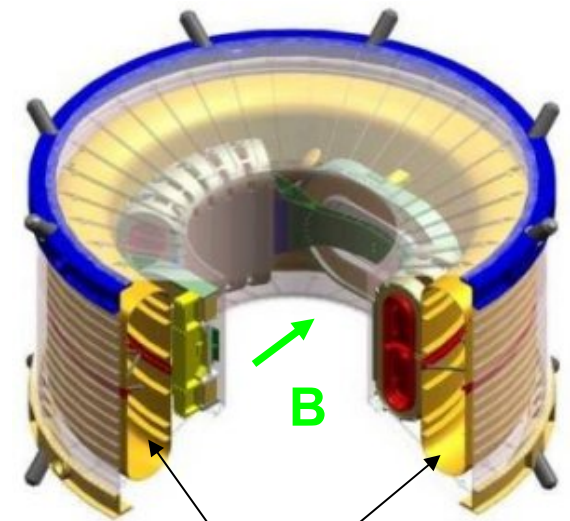
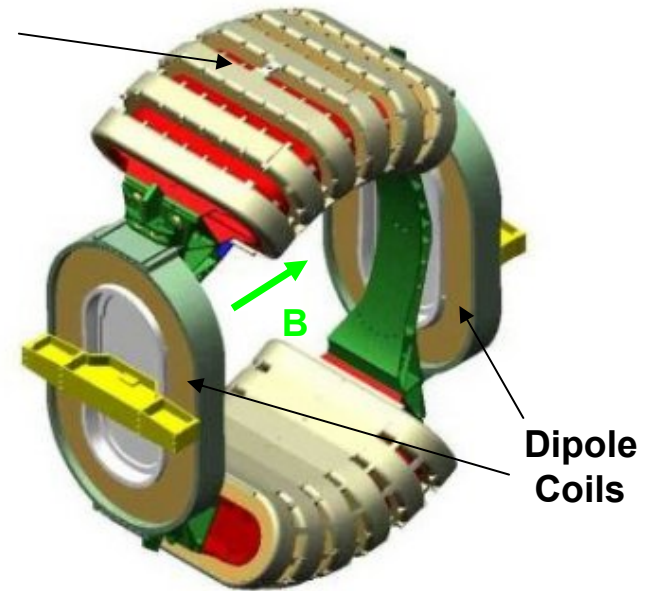
AMS01 SEEN FROM MIR IN THE CARGO BAY OF DISCOVERY

AMS02 Superconducting Magnet



- 3000 Liters Superfluid He
- cryocoolers
- $BL^2 = 0.8 \text{ TM}^2$
- Dipole field inside, no Dipole moment

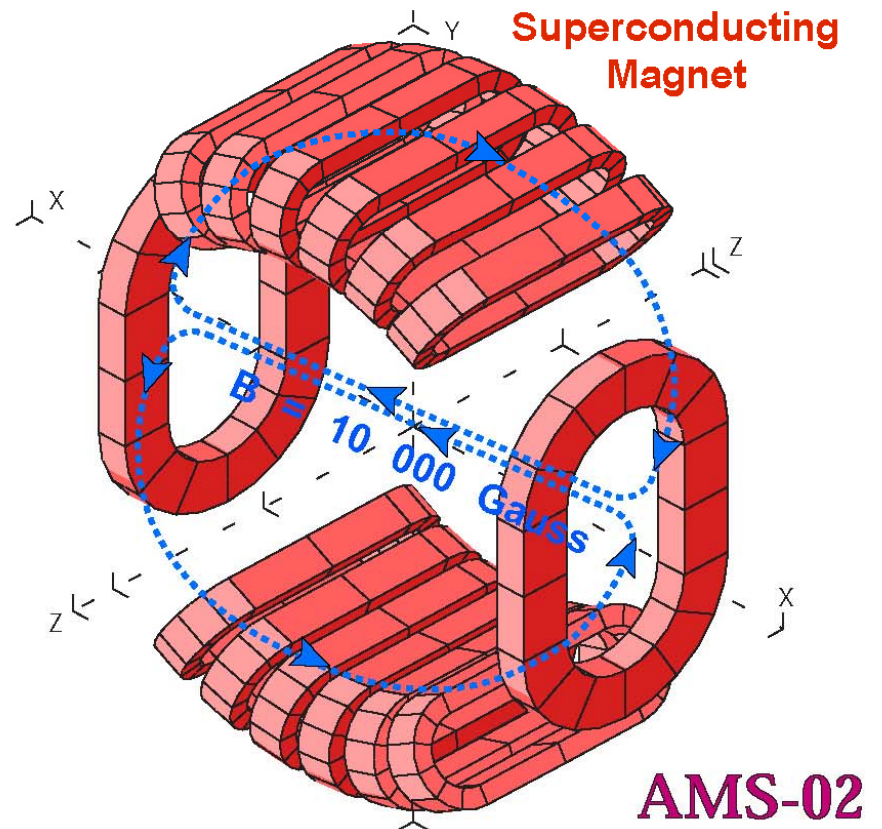
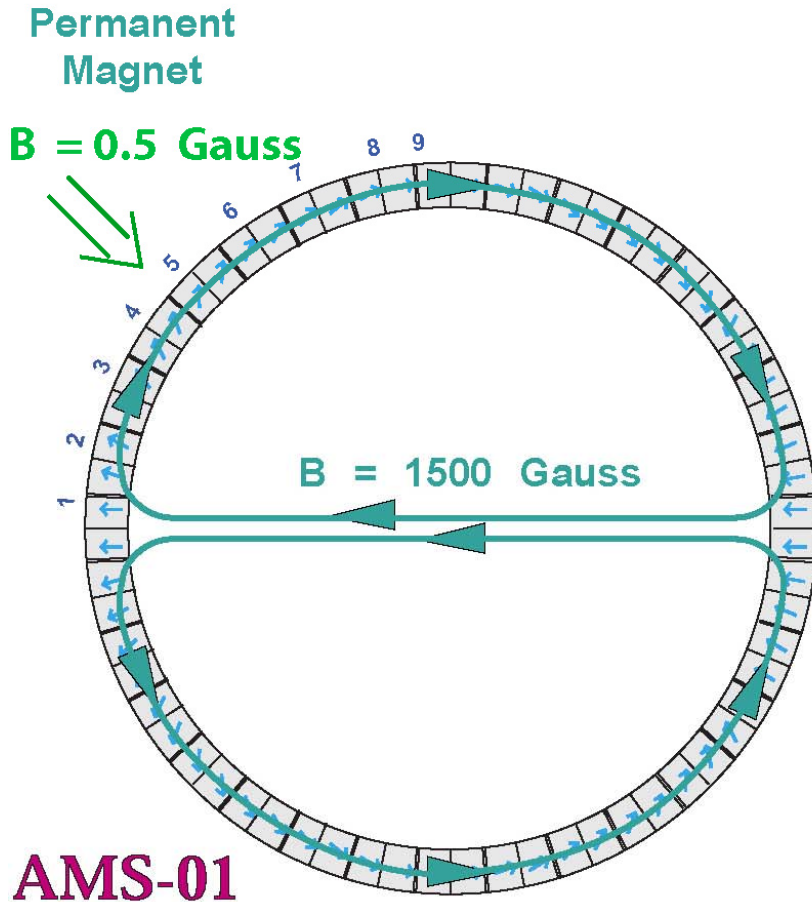
Flux Return Coils



He Vessel

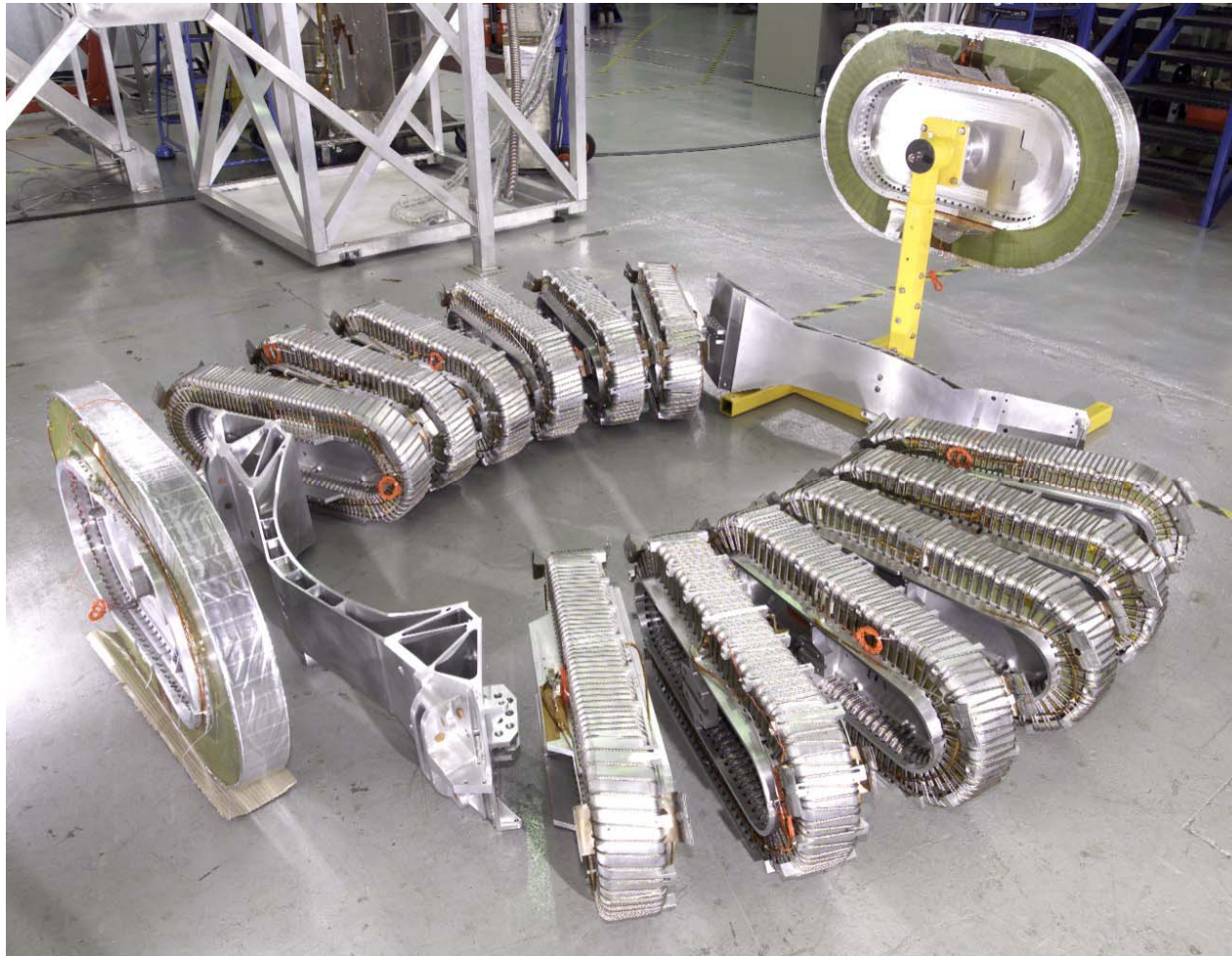
Identical configuration between AMS-01 and AMS-02

1. Same large volume ($\Phi = 3'$, $h = 3'$)
2. No Dipole momentum (no rotation in earth mag. field)
3. No quench



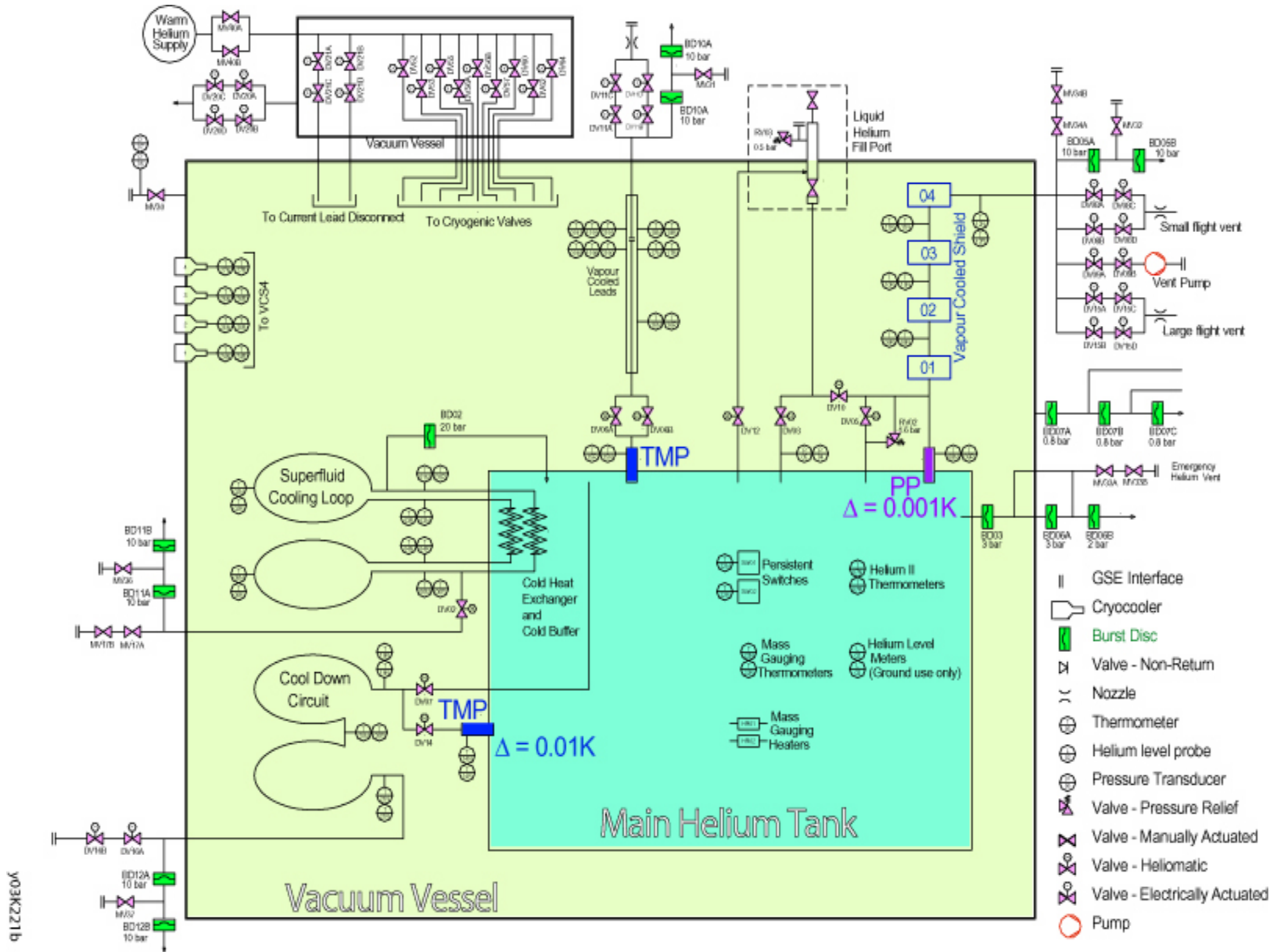
AMS02 Superconducting Magnet

2 Dipoles coils and 12 racetrack coils



**ALL THE COILS
ARE BUILT AND
TESTED.**

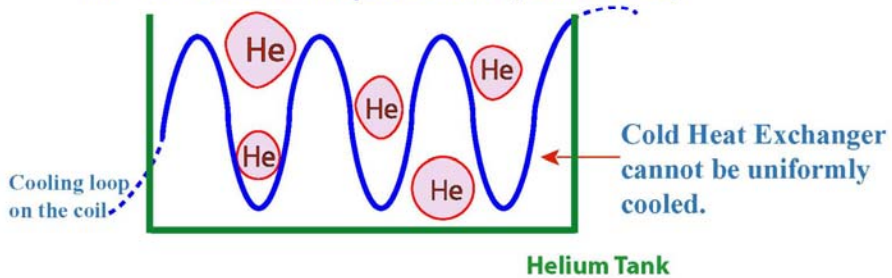
AMS02 Superconducting Magnet cooling system



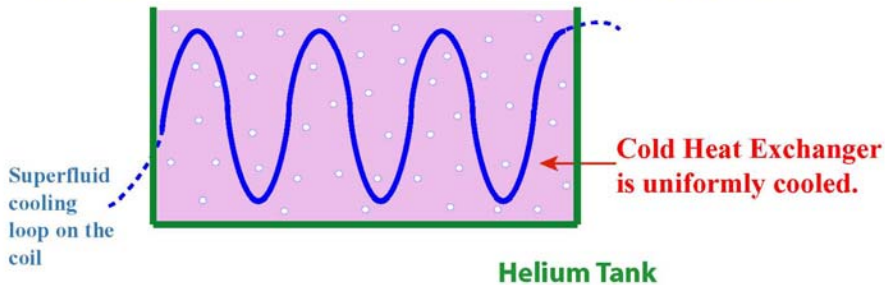
COOLING WITH SUPERFLUID HELIUM

Liquid Helium in Space

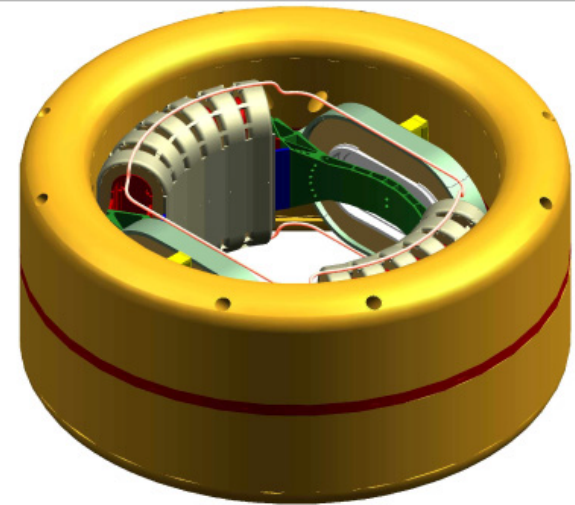
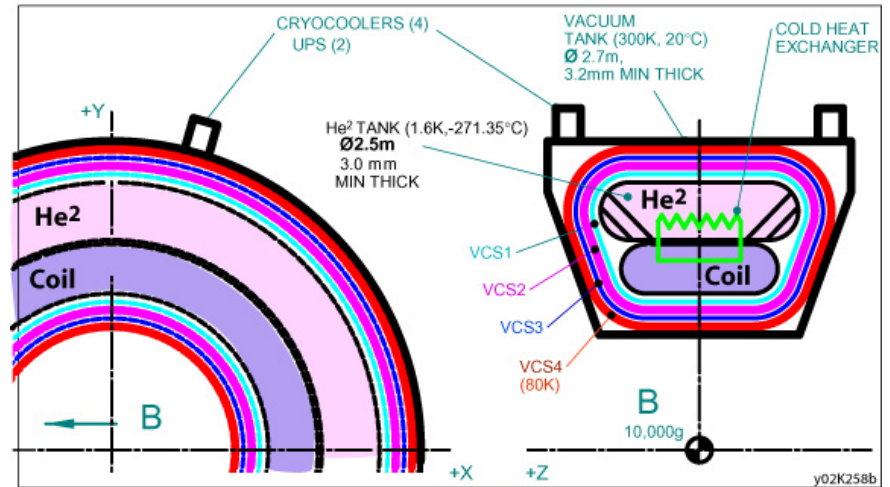
He¹ : Normal Liquid: 4.3 K (-268.85 C)



He² : Superfluid Helium: 1.6 K (-271.35 C)
is a Bose-Einstein Liquid; it has no surface tension



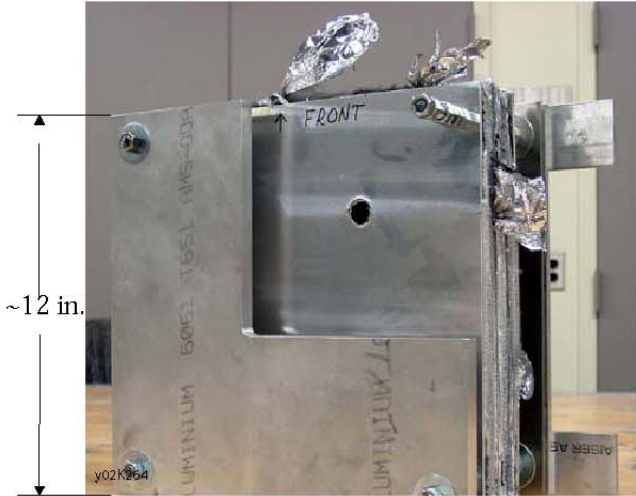
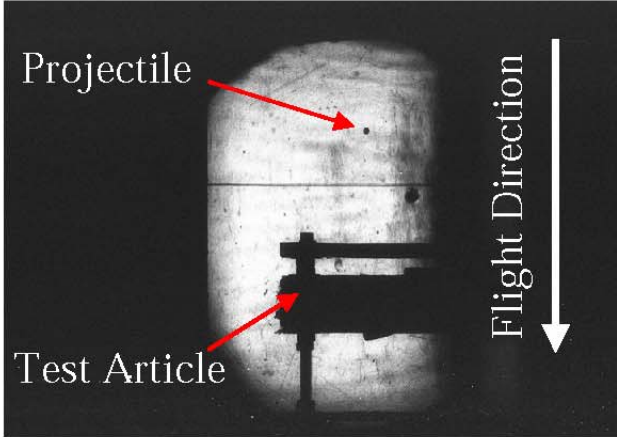
Superfluid helium vessel



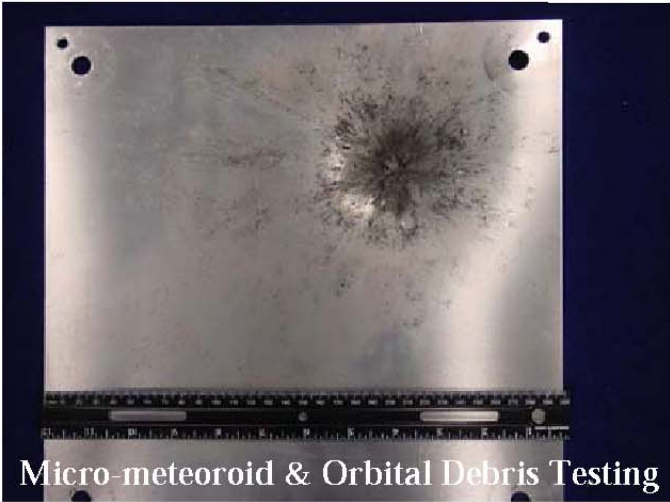
A puncture in the tank would result in **expanding** the volume from 2500 litres to 1.9×10^6 litres (six times the volume of the shuttle cargo bay) with extremely serious consequences.

y02K121gHarrison

Micro-meteoroid & Orbital debris testing

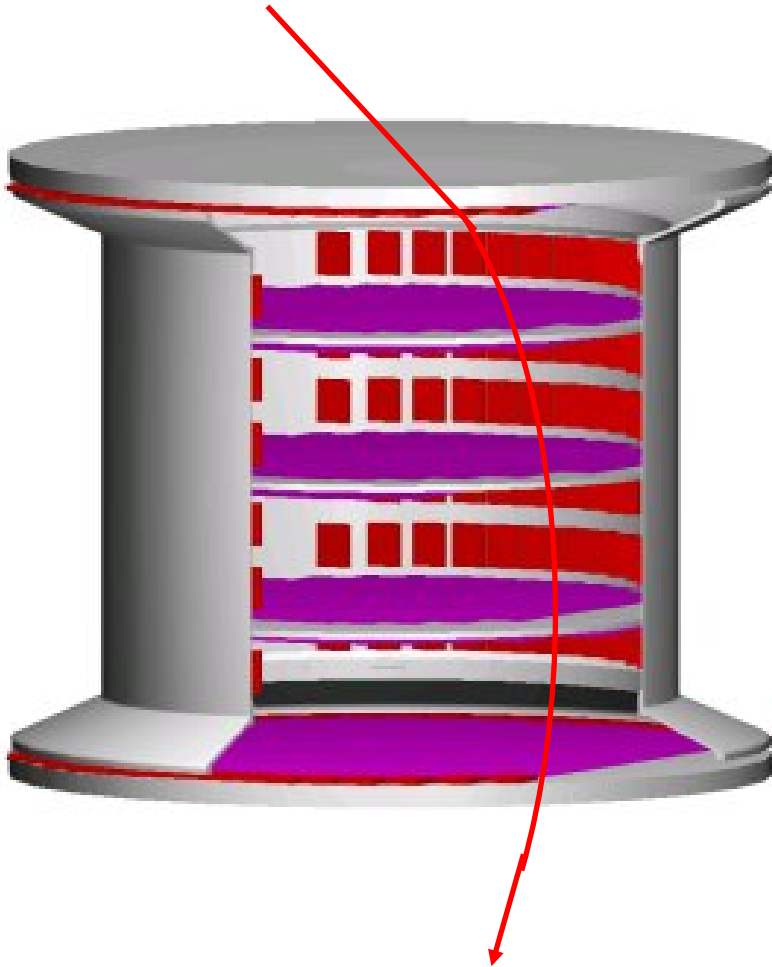


Hole made in Vacuum Case Layer from 5mm Aluminum Orbital Debris Particle shot at ~ 15,000 mph



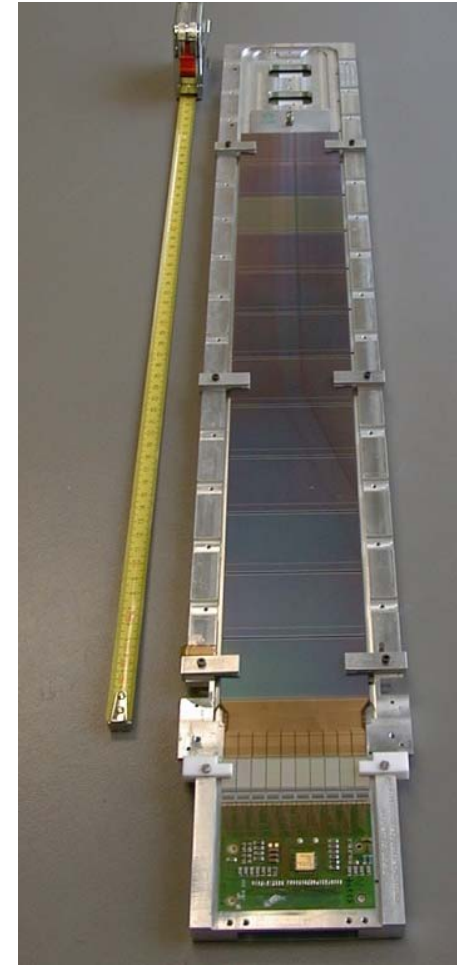
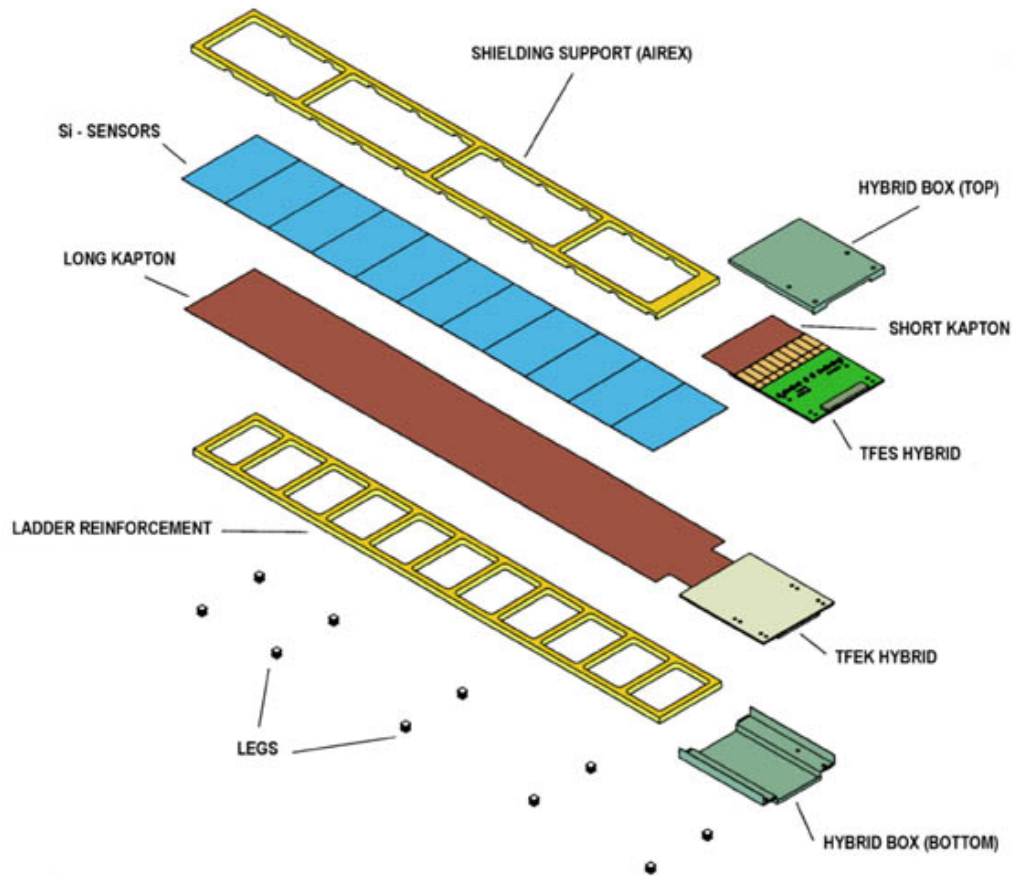
Damage to SFHe tank due to 5 mm Al Orbital Debris Particle. Tank not punctured

Silicon Tracker



- **8 Double-sided planes, $\sim 6\text{m}^2$**
- **Pitch:**
 - Bending $27.5\ \mu\text{m}$ (**coord res $10\ \mu\text{m}$**)
 - Non-Bending $104\ \mu\text{m}$ (**coord res $30\ \mu\text{m}$**)
- **Rigidity: $dR/R \approx 2\%$**
for 1-10 GeV protons with magnet
- **Signed Charge (dE/dx)**

Silicon Tracker Ladder



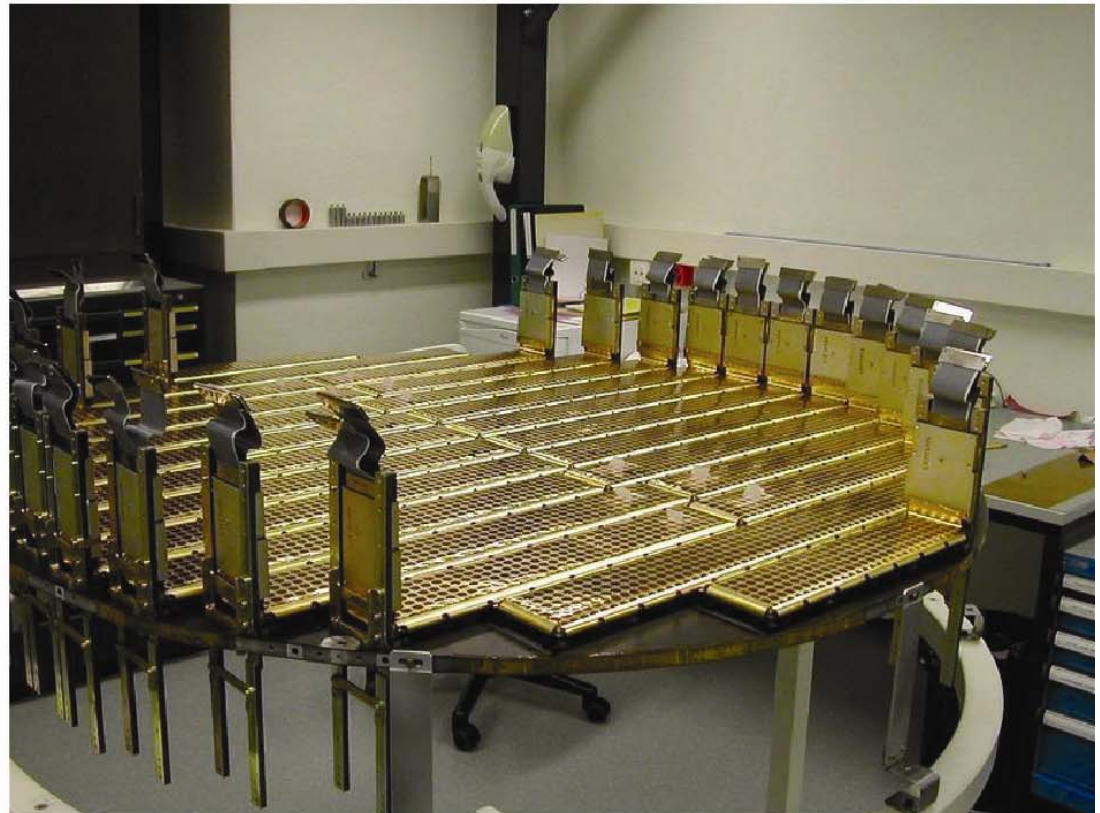
Silicon Tracker Construction



Sensor Positioning to a few μ accuracy

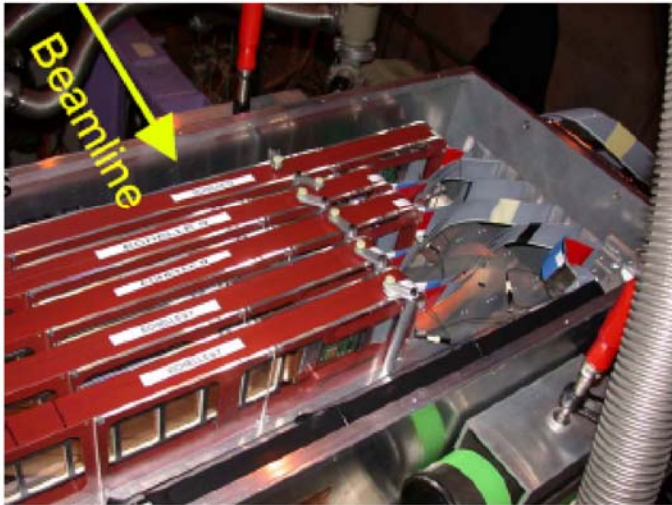


Bonding and inspection of ladders

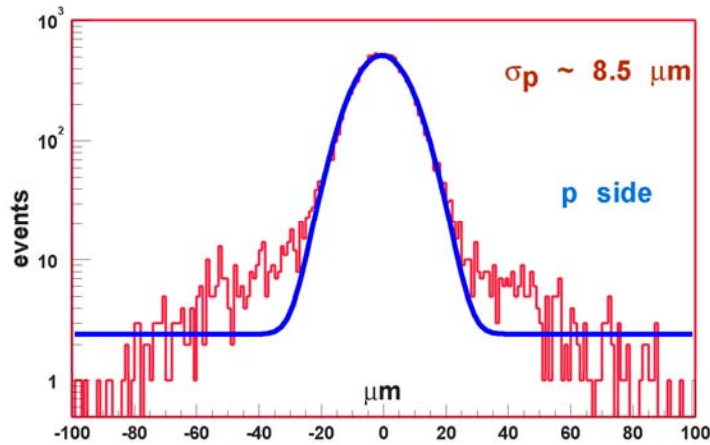


Tracker integration : layer L2 on plane P2

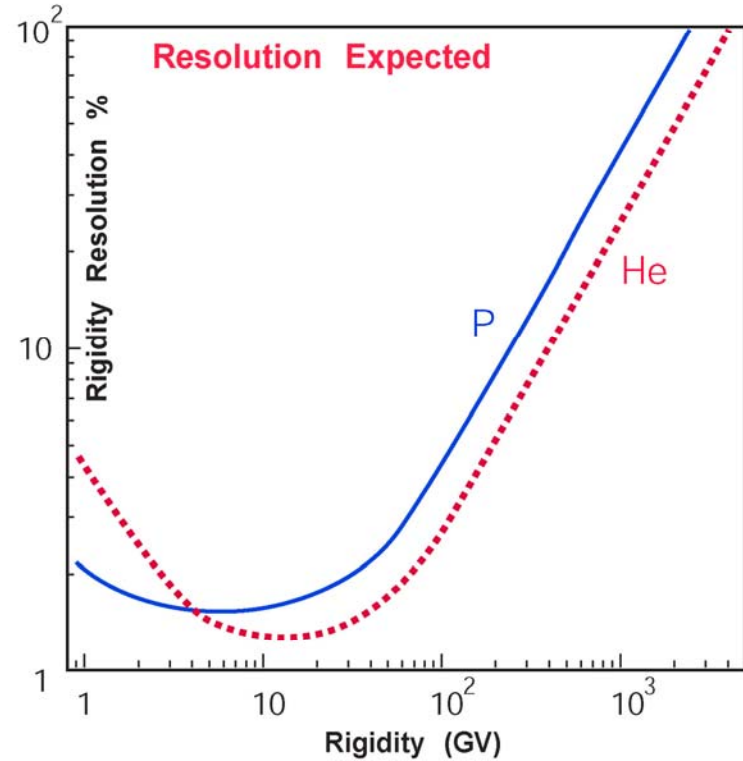
Performances of the Silicon tracker



Ladders in a test beam at CERN

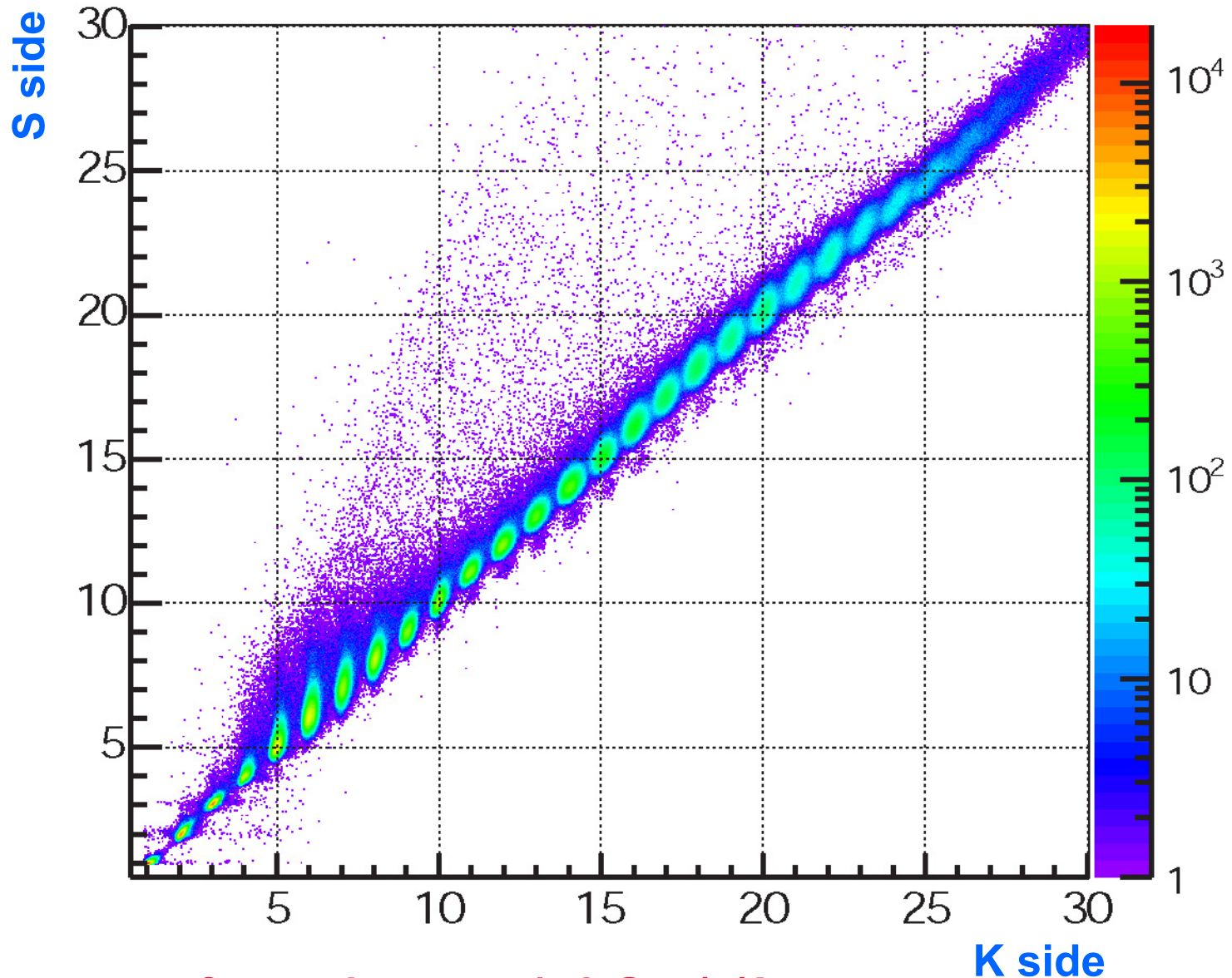


Test of ladders in a 120 GeV/c muon beam.
Hits with respect to the position from the beam



Calculated Rigidity resolution for p and He

Charge Measurement with Silicon Tracker

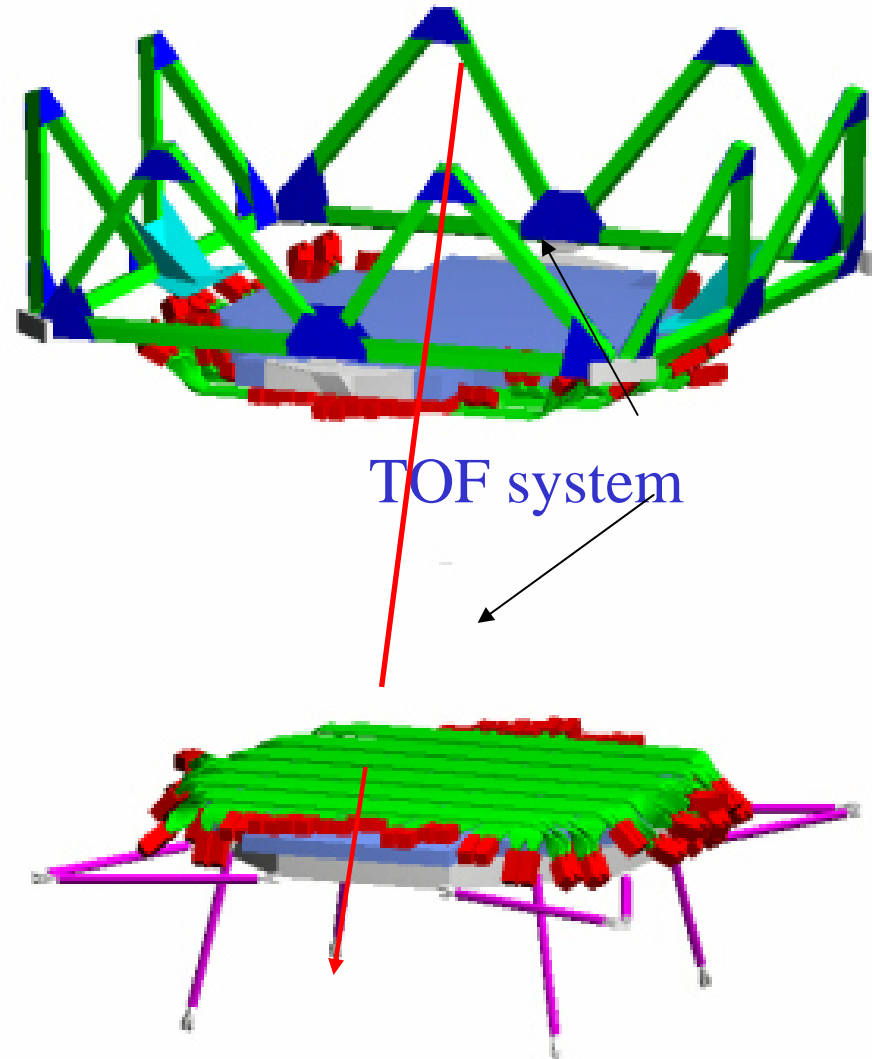


Fragments from a beam at 158 GeV/c/A

Time-of-flight system

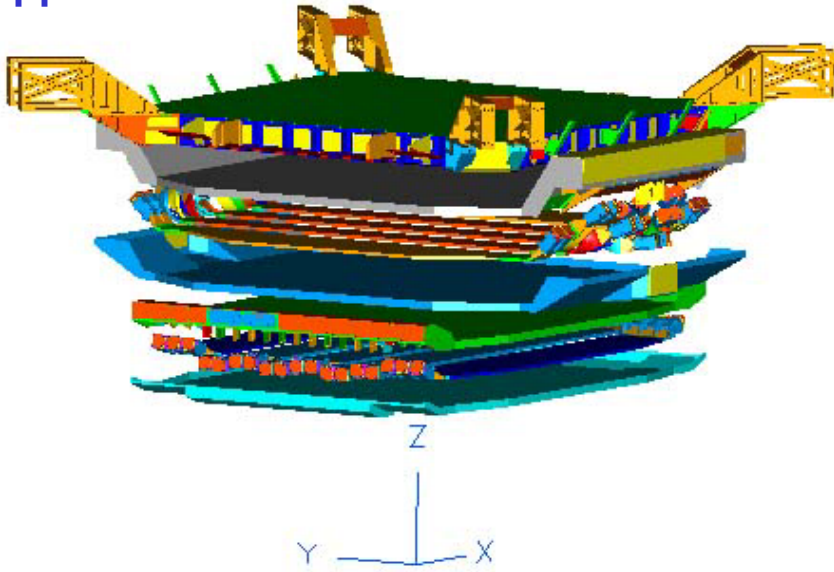
(See talk C.Sbarra, this conf.)

- Time-of-flight (**velocity**).
- Charge Determination (**dE/dx**)
- Fast Trigger
- Up/Down Separation
- Scintillator Paddles With Phototubes at Both Ends
- **120 ps Time Resolution**
- 8 m² Total Area
- 4 Planes (2 upper, 2 lower)



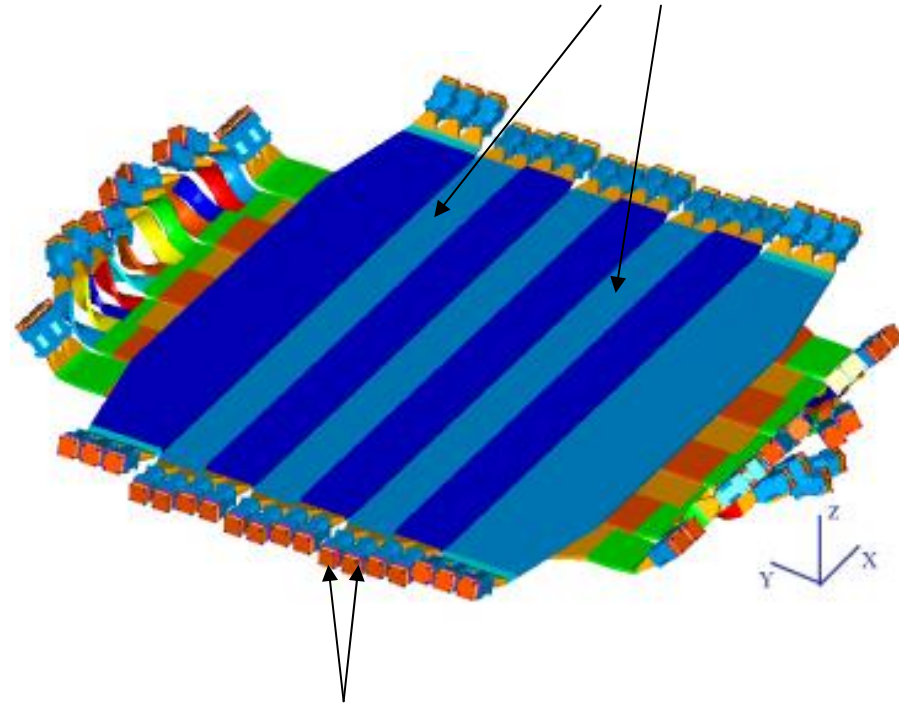
STRUCTURE OF THE TOF COUNTERS

Upper TOF

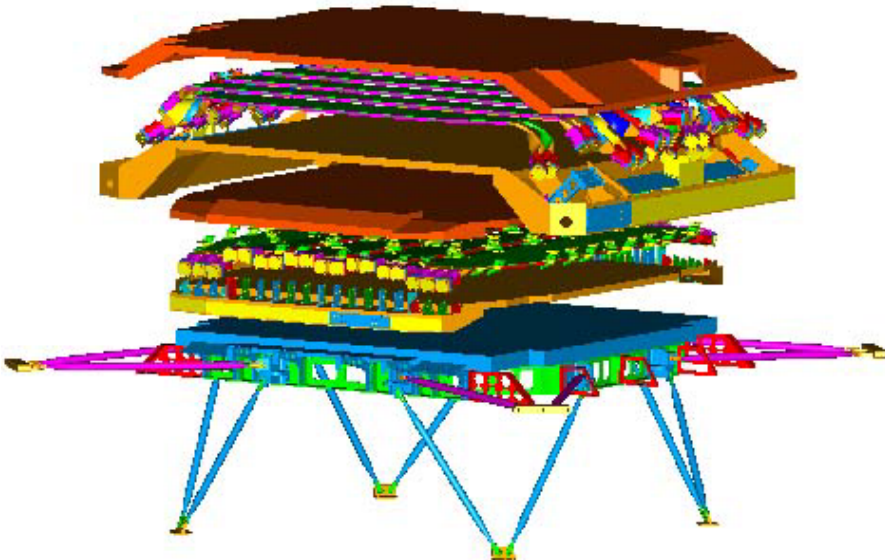


Complicated light guides structure to cope with high magnetic field

Scintillator Paddles



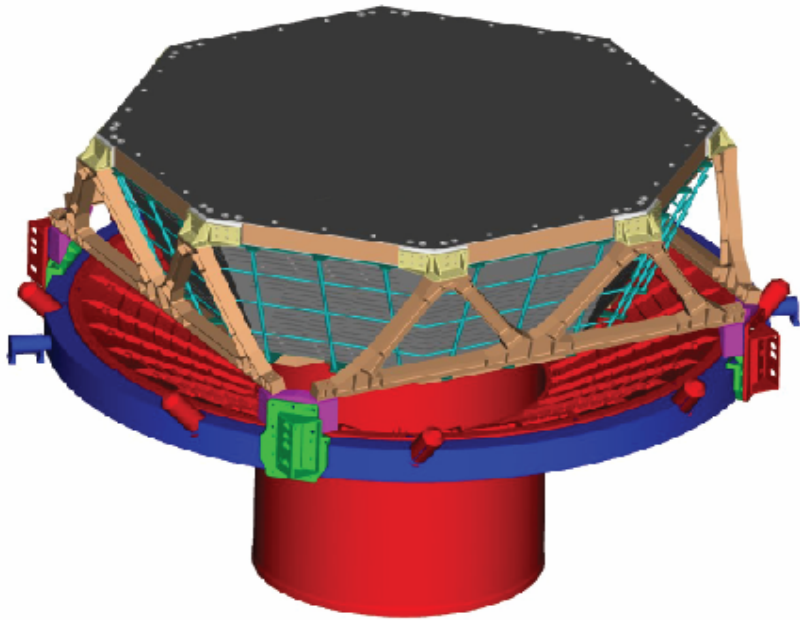
Lower TOF



Dual Photomultipliers for Redundancy and time resolution

Transition Radiation Detector (TRD)

(See talk by T. Siedenburg. This conf.)



20 layers

328 modules each consisting of :

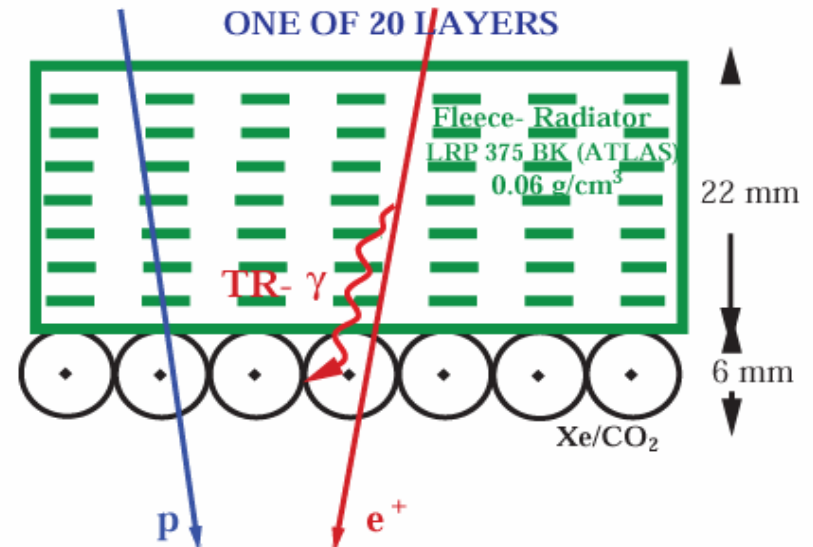
22mm fleece

16 - 6 mm straw tubes

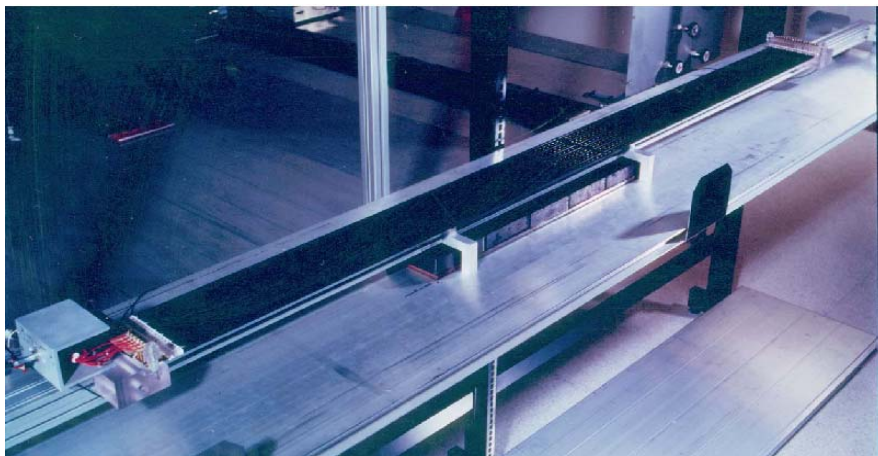
(Xe/CO₂ – 80/20)

- **e⁺/e⁻ rejection $10^2 - 10^3$**
in 1.5 – 300 GeV range

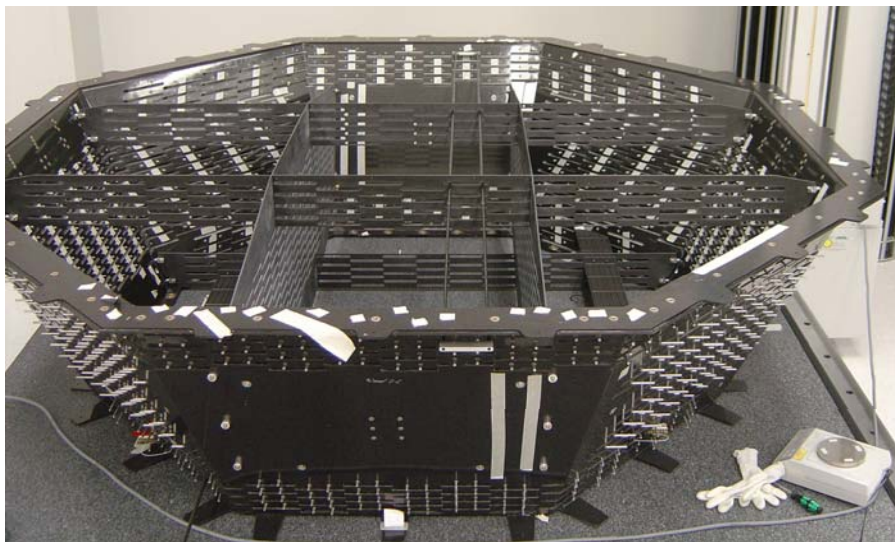
- with ECAL:
e⁺/p rejection **$>10^6$**



Transition Radiation Detector Construction

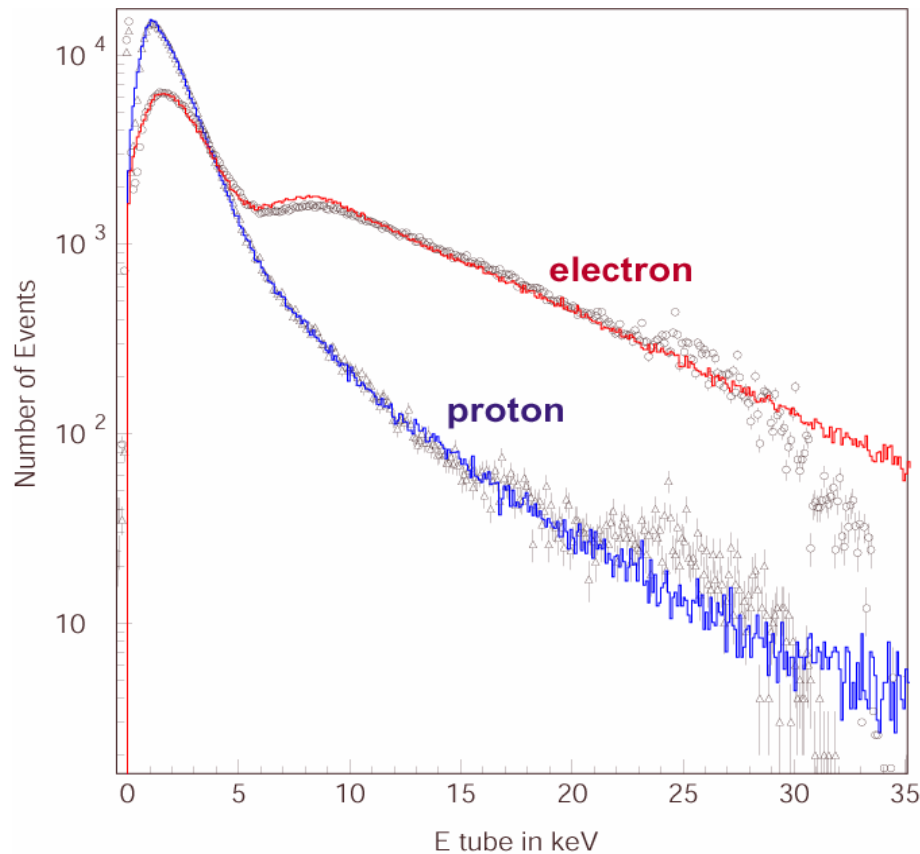


1.5 meter module on assembly jig



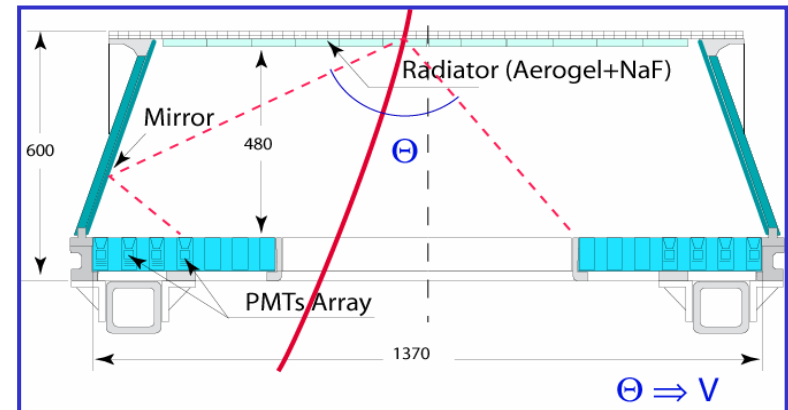
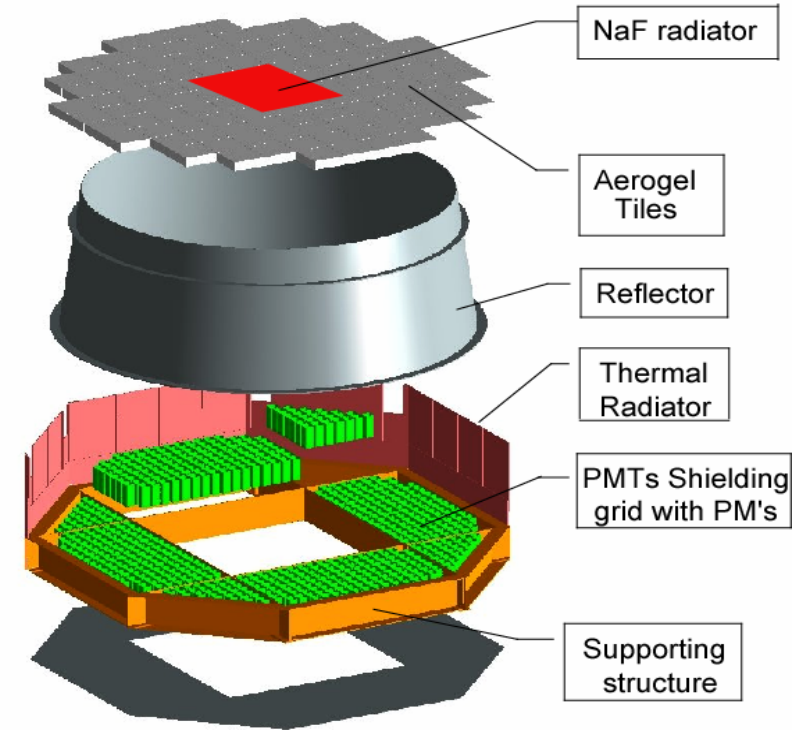
Carbon fiber composite/Aluminum Honeycomb
Mechanical support structure

Test at CERN with protons and electrons
in 5 to 250 GeV range

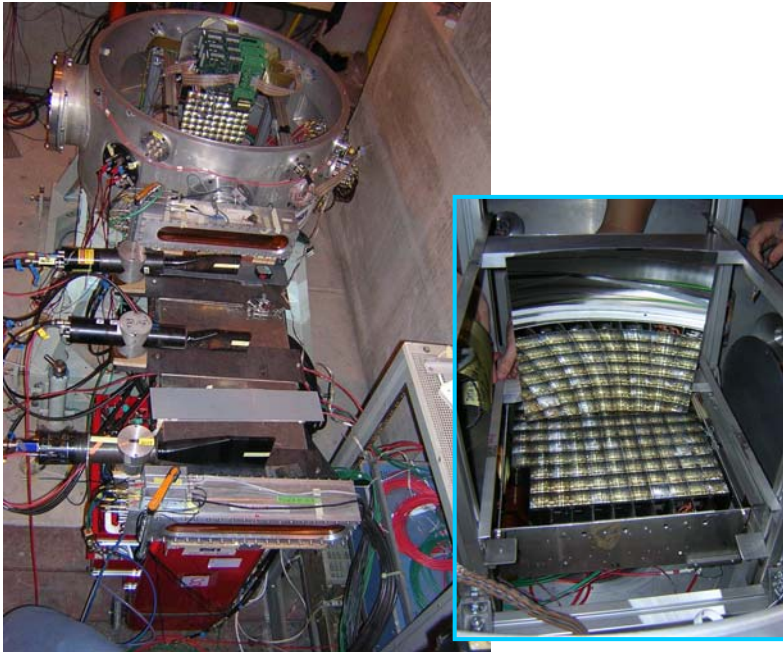


Ring Imaging Čerenkov Counter

- **Z** measurement
- Precise measurement of velocity $\delta\beta/\beta \sim 0.1\%$ from Čerenkov cone opening angle
- Combined with p from tracker, allow isotopic separation of nuclei up to $A < 25$ and $p < 12 \text{ GeV}/c/N$
- Albedo rejection
- Dual solid radiator configuration Aerogel ($n=1.035$, 3cm thick, threshold 3 GeV/c) and NaF ($n=1.33$, 0.5cm thick, threshold 1 GeV/c) to extend energy range
- High accuracy mirror to improve light collection
- 680 PMT's multianodes (16 pixels). Granularity $(8.5\text{mm})^2$

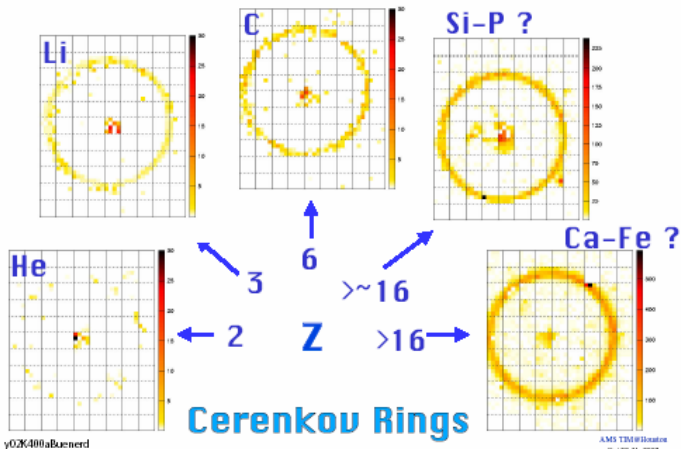
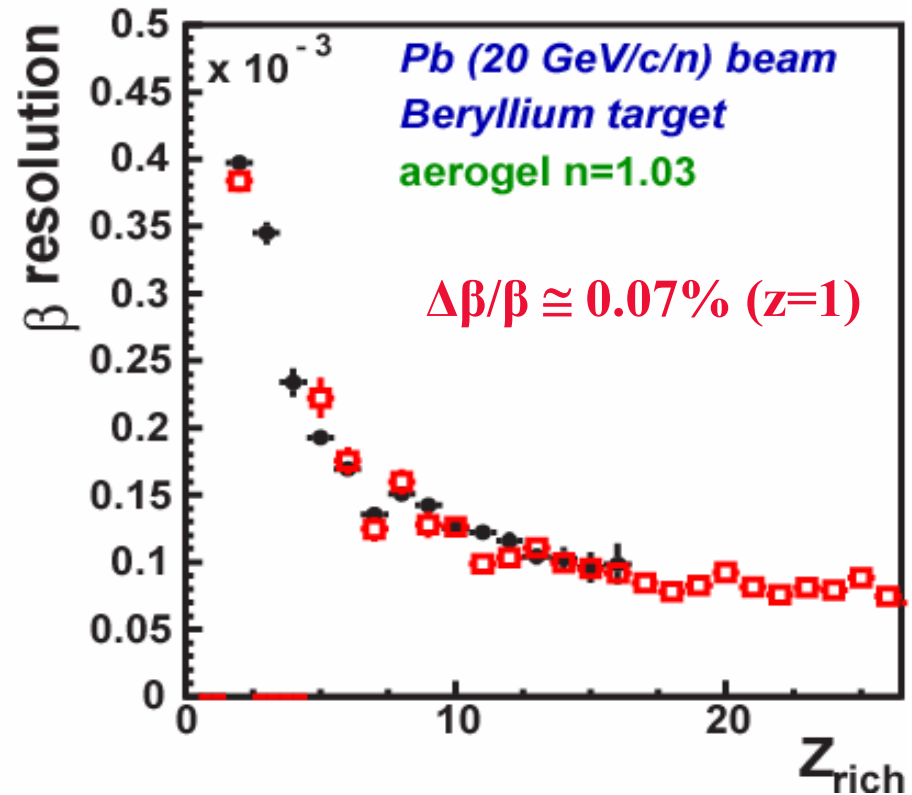


RICH PERFORMANCES (TEST BEAM)



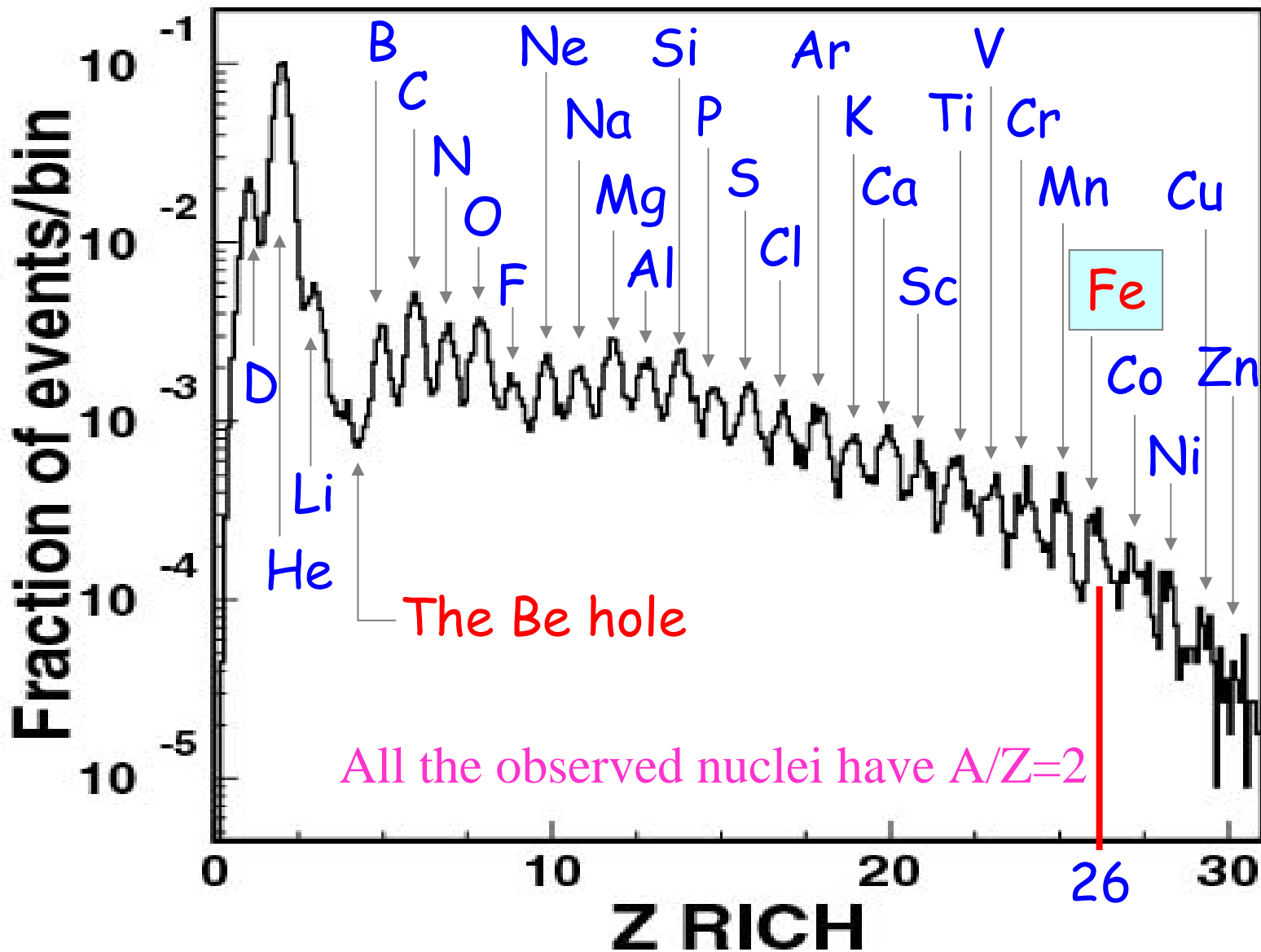
Experimental Setup in the Test beam and closeup view of the mirror system and PM's

Resolution on velocity
for nuclei up to Z=25



Raw data from test beam

RICH reconstructed charge spectrum

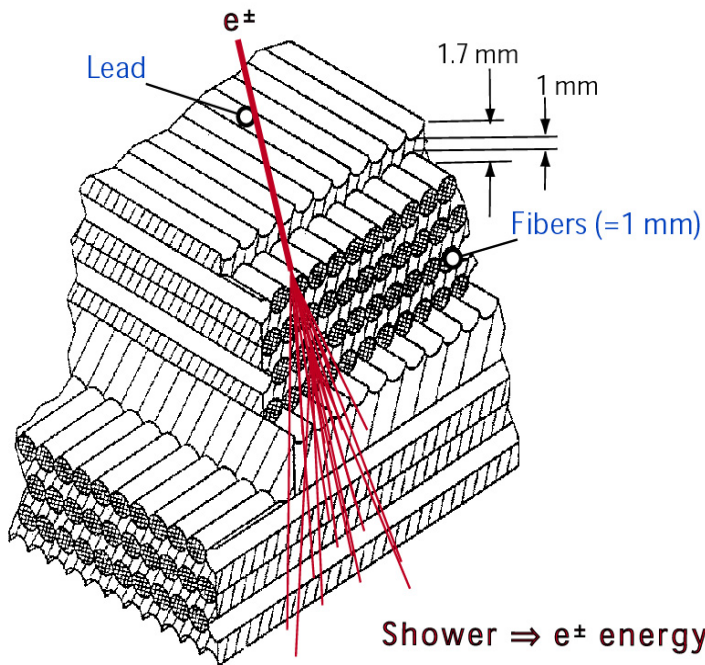


3D IMAGING ELECTROMAGNETIC CALORIMETER (ECAL)

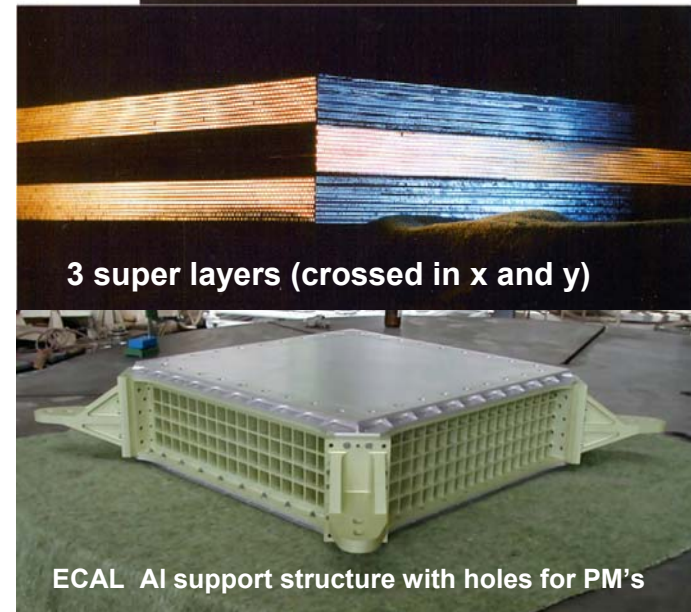
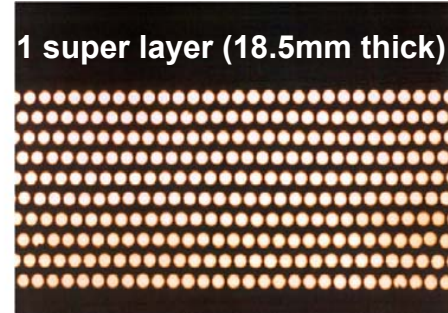
Lead/fibers sandwich
9 super layers

(size 65.8x65.8cm², thick 18.5mm each)

16.5 X₀ (X₀ ≅ 1 cm)
18 samplings in depth
Lateral granularity 0.9 cm
(0.5 moliere radius)



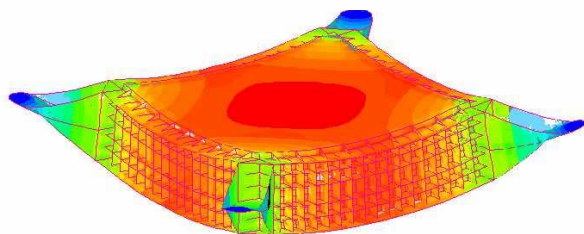
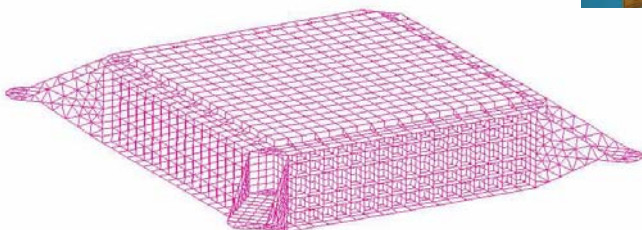
Measures energy and angle of γ , e^+ , e^-
from 1 GeV up to a few TeV



10⁻³ p^\pm Rejection at 95% e^\pm Efficiency Via Shower Profile

SPACE QUALIFICATION TESTS. BEIJING. 01/2003

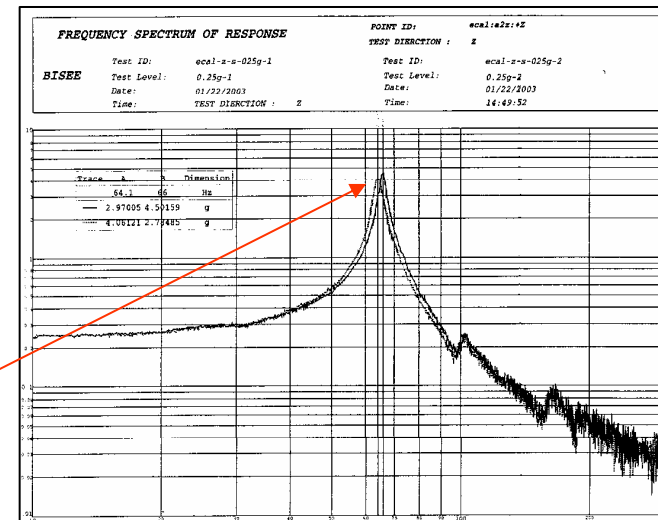
FEA Model



*Calculation :
1st vibration mode at 69 Hz*



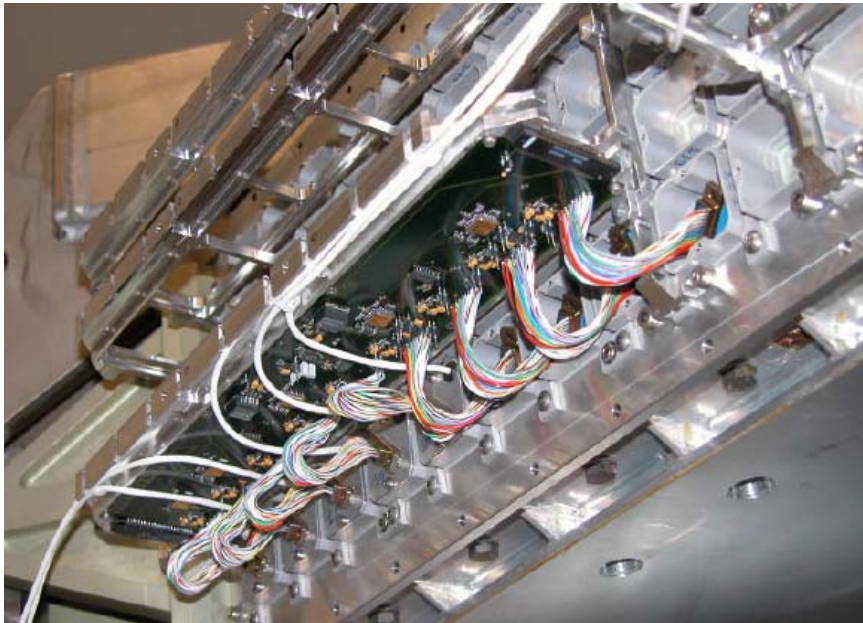
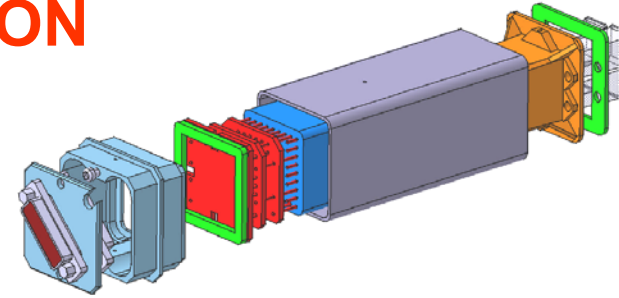
**First eigenfrequency
At 65 Hz**



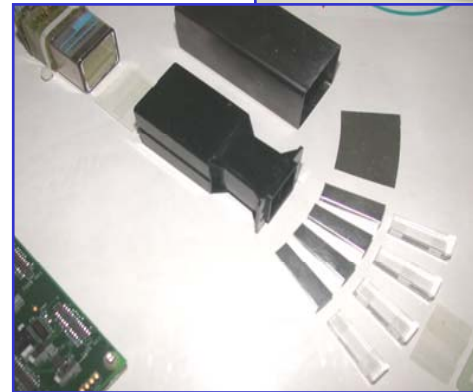
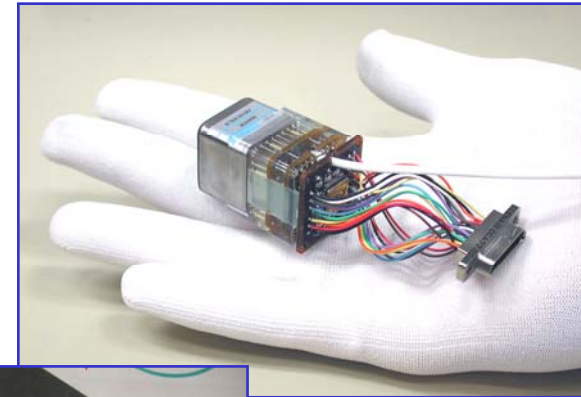
**ECAL AND MODEL
VALIDATED**

ECAL INSTRUMENTATION

- Fibers read by 324 4-pixels pmt's
- FE readout electronics with ASIC
- 60 000 dynamic range
- Stand alone γ trigger for physics of high energy cosmic gamma rays

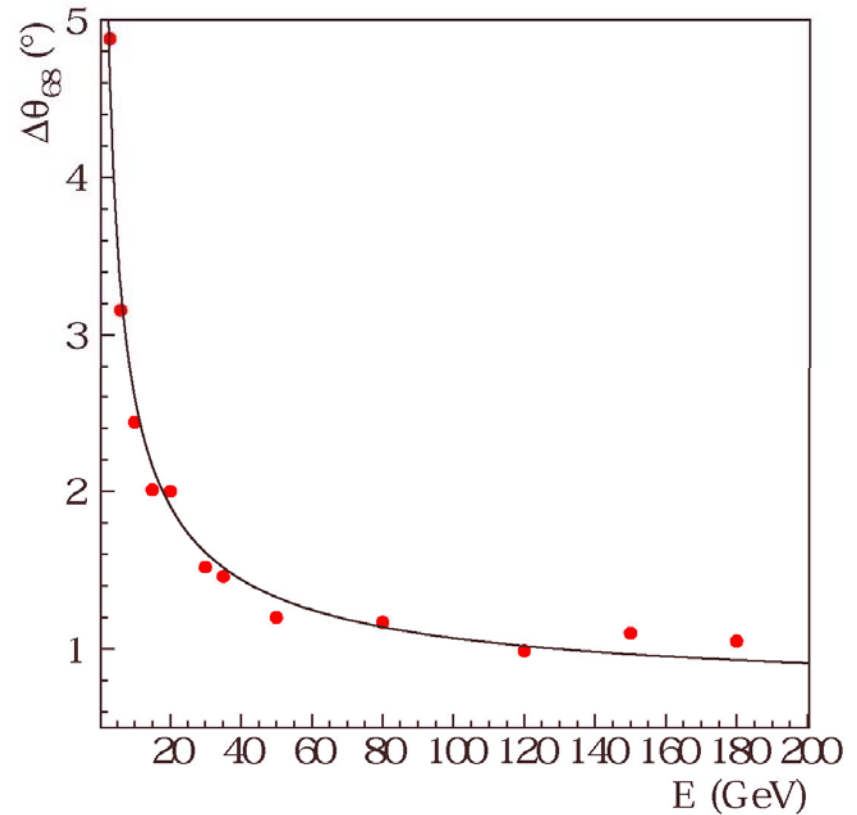
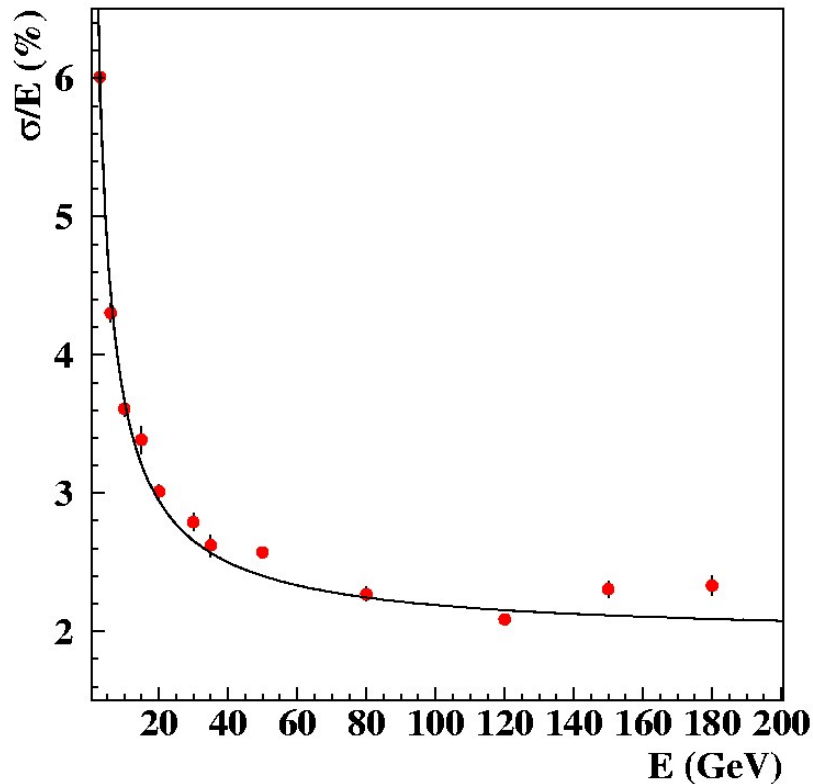


ECAL equipped for test beam
Space qualification in Terni



Light collection system with
Front end electronics

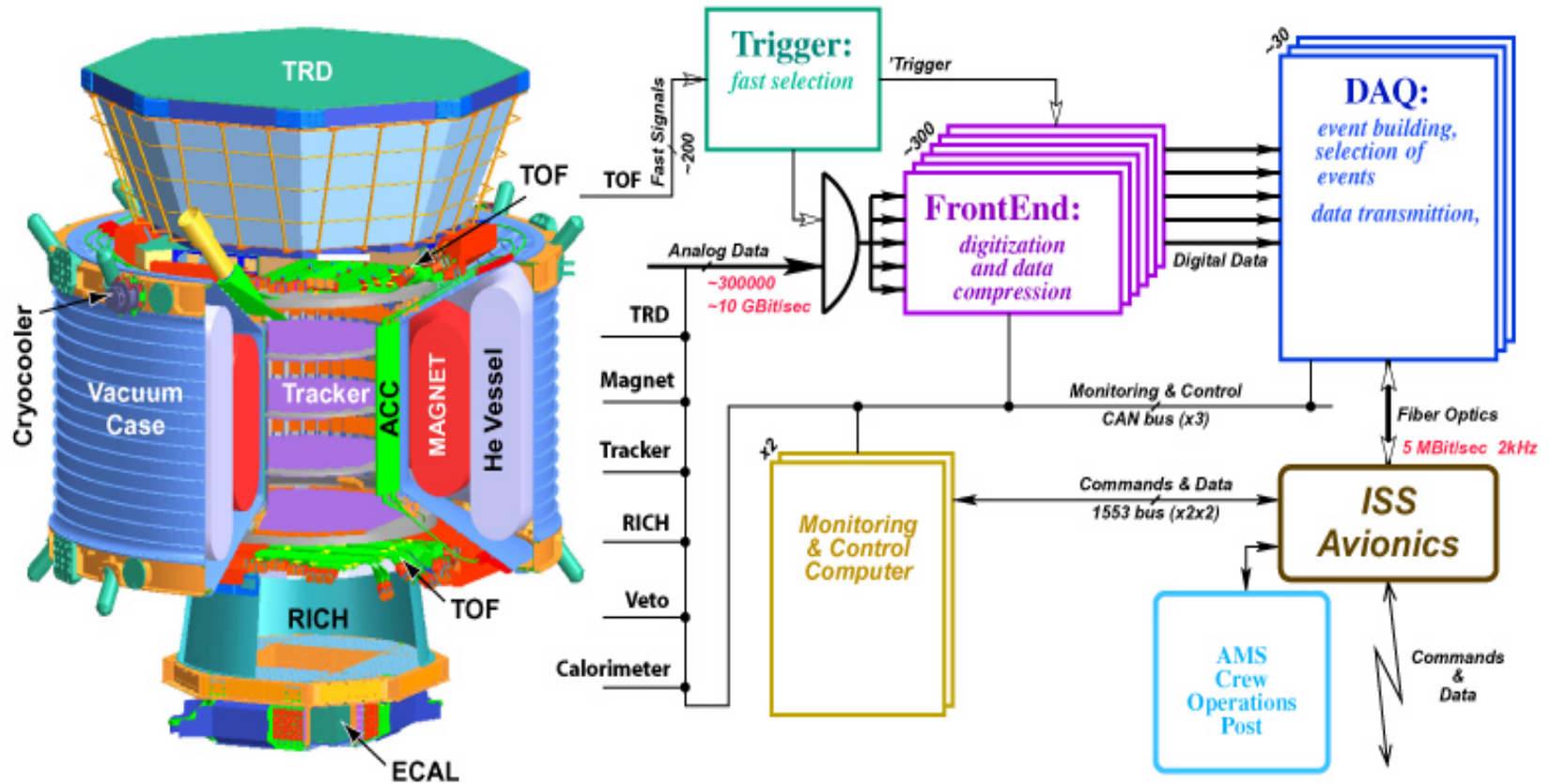
ECAL ENERGY AND ANGULAR RESOLUTION (FROM BEAM)



$$\sigma/E = (10.89 \pm 0.28)/\sqrt{E} \oplus (2.48 \pm 0.05) \%$$

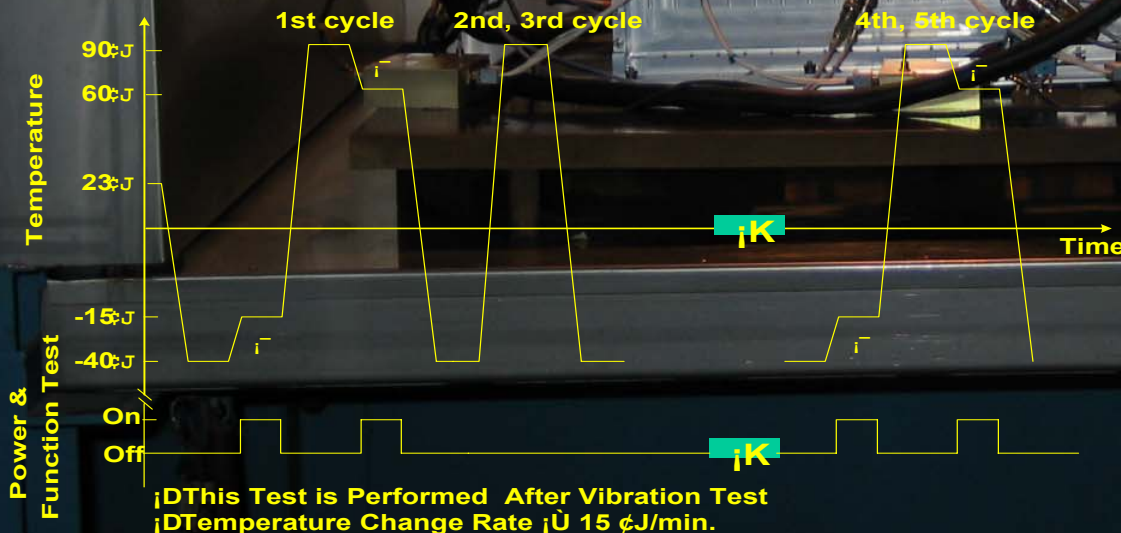
$$\Delta\theta_{68} = (8.06 \pm 0.11)/\sqrt{E} \oplus (0.60 \pm 0.04)^\circ \text{ (at 68 \% c.l.)}$$

Overview of AMS-02 Electronics on the ISS



J-Crate in Thermal Chamber

M.Capell



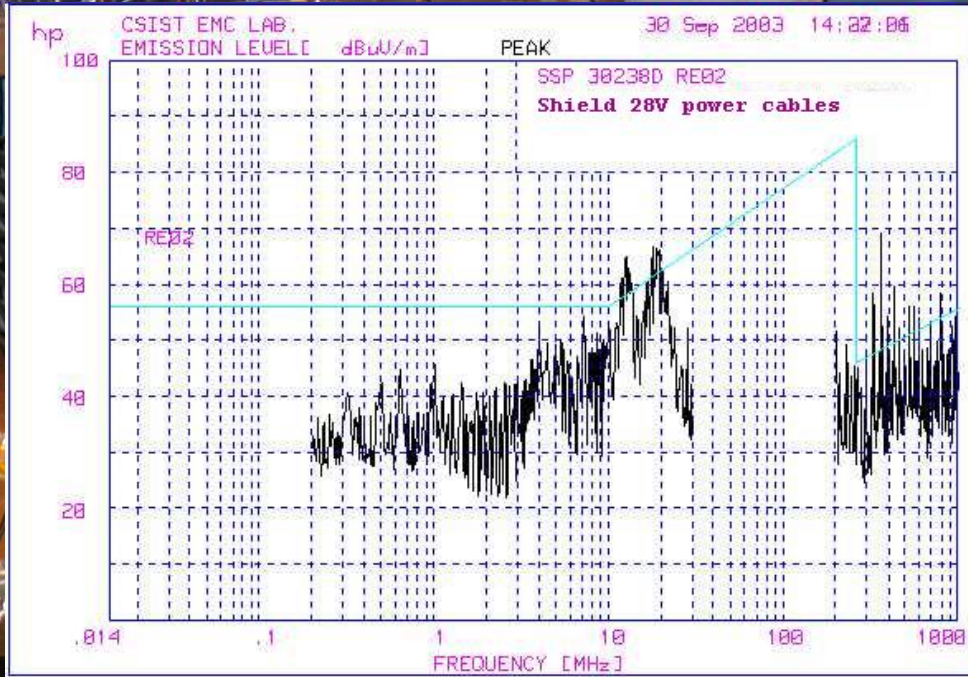
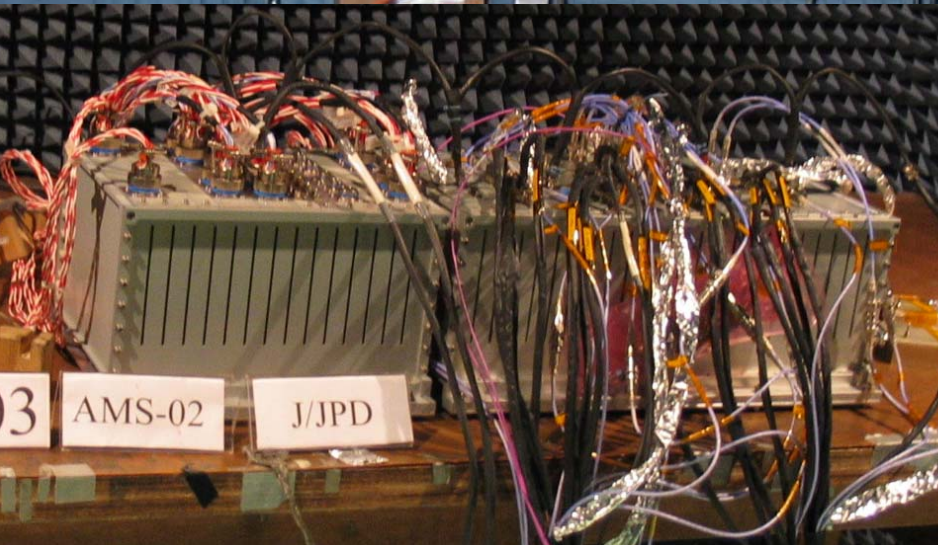
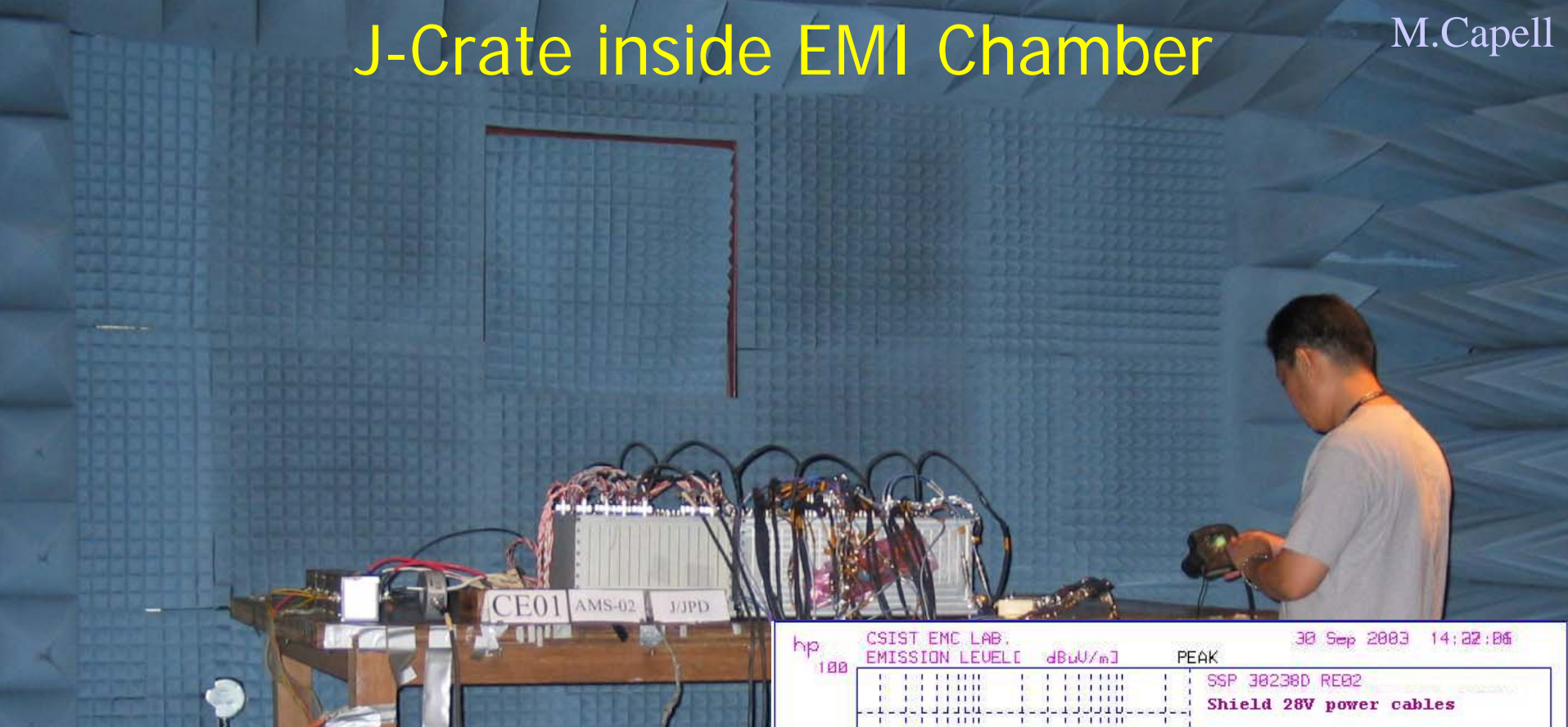
Thermal
Qualification

Operating: -15, +60C

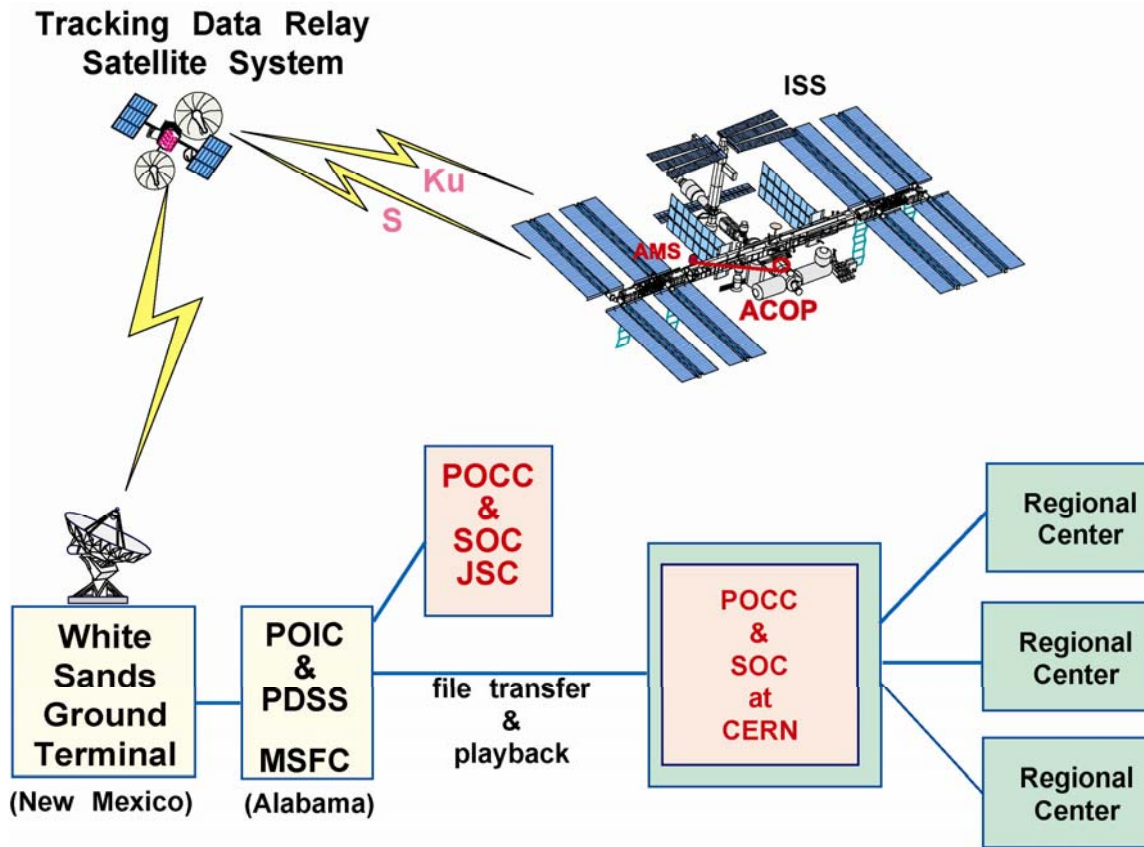
Non-Oper: -40, +90C

iDThis Test is Performed After Vibration Test
iDTemperature Change Rate iU 15 ¢J/min.

J-Crate inside EMI Chamber



AMS DATA TRANSFER AND PROCESSING



2 Mbits/sec average science data downlink rate

ACOP (AMS Crew Operation Post) allows ISS local storage and backup of data when downlink is off.

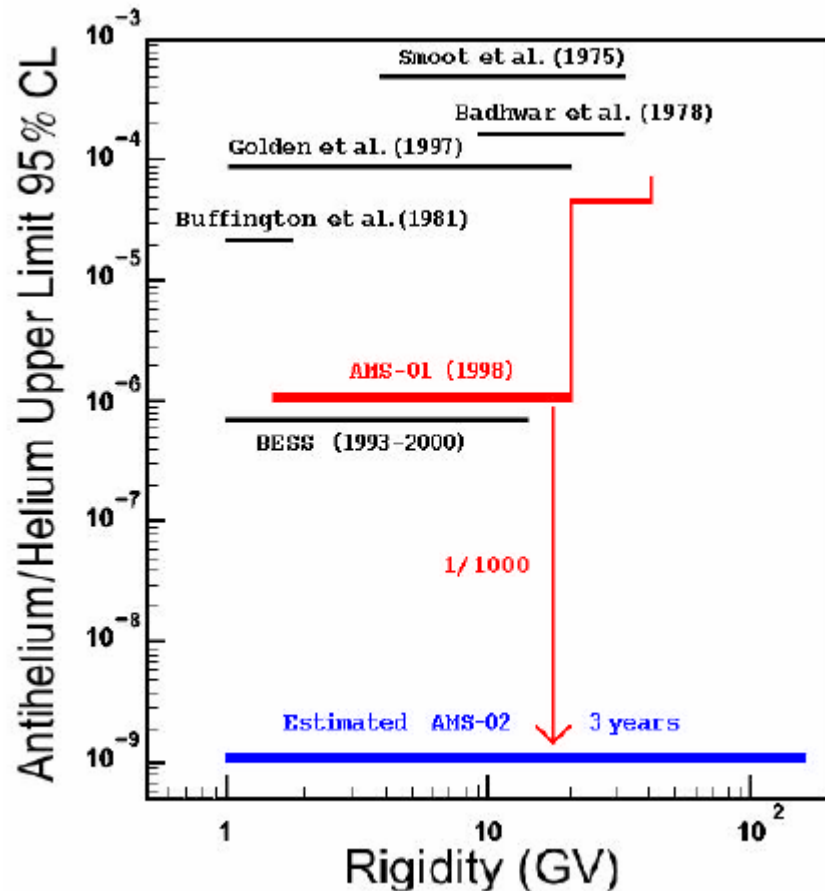
Main processing center at CERN

ACOP	AMS Crew Operation Post
POCC	Payload Operation Control Center
SOC	Science Operation Center
MSFC	Marshall Space Flight Center (Al)
JSC	Johnson Space Flight Center (Tx)

Physics performances of AMS-02

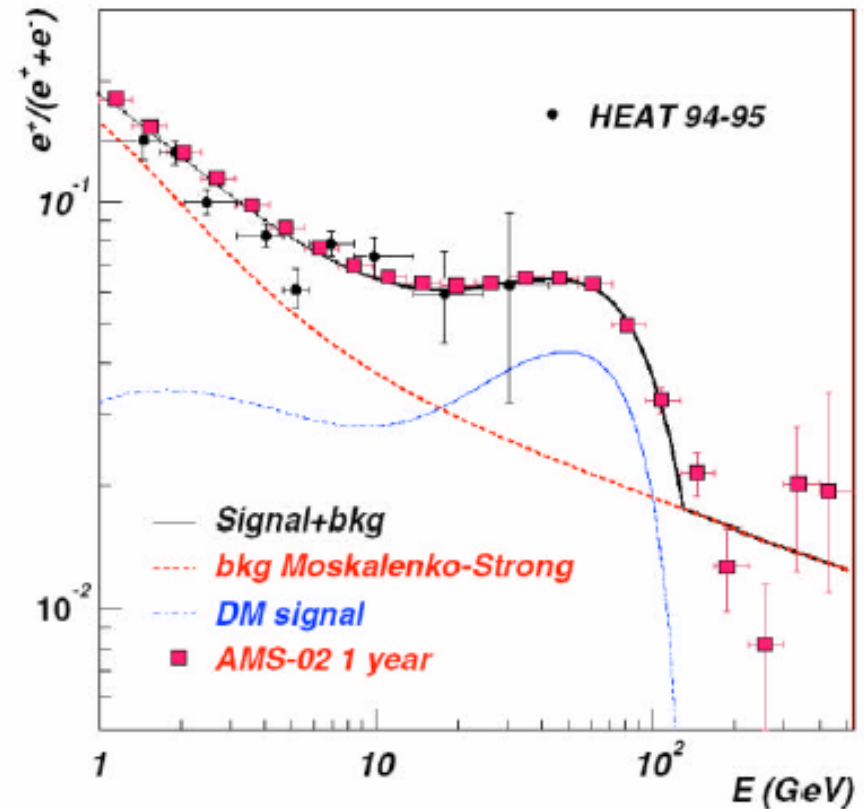
PRIMORDIAL ANTIMATTER

Limit on antihelium
1 year data taking



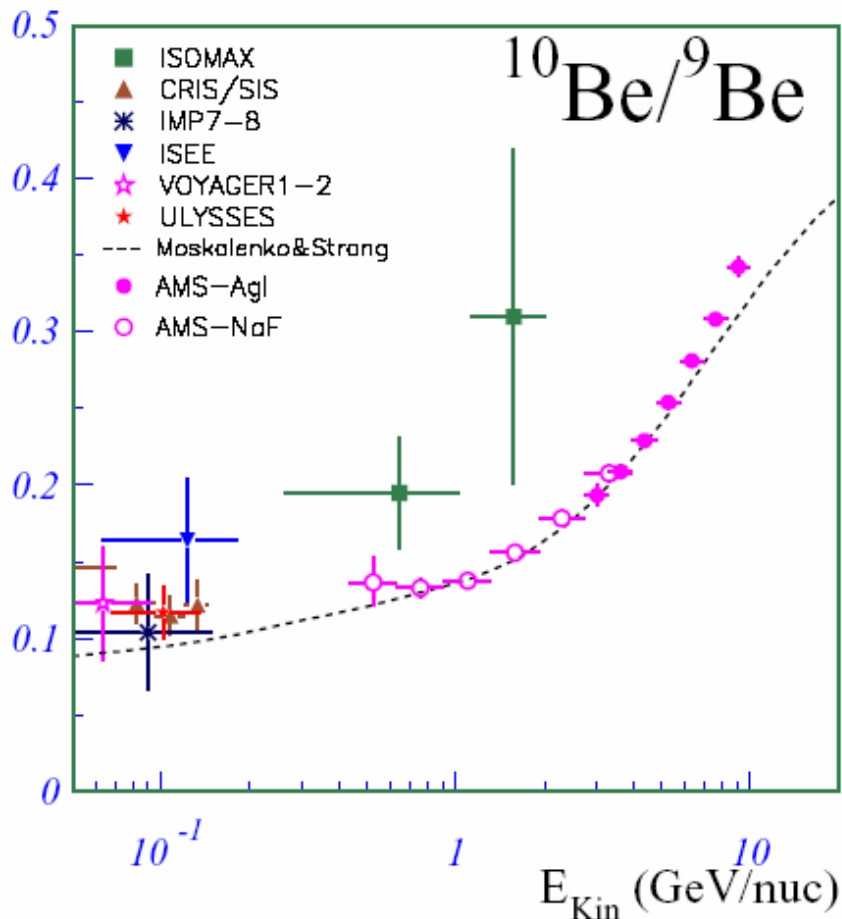
DARK MATTER

Positron spectra with neutralino
of mass 130.3 GeV/c²
1 year data taking

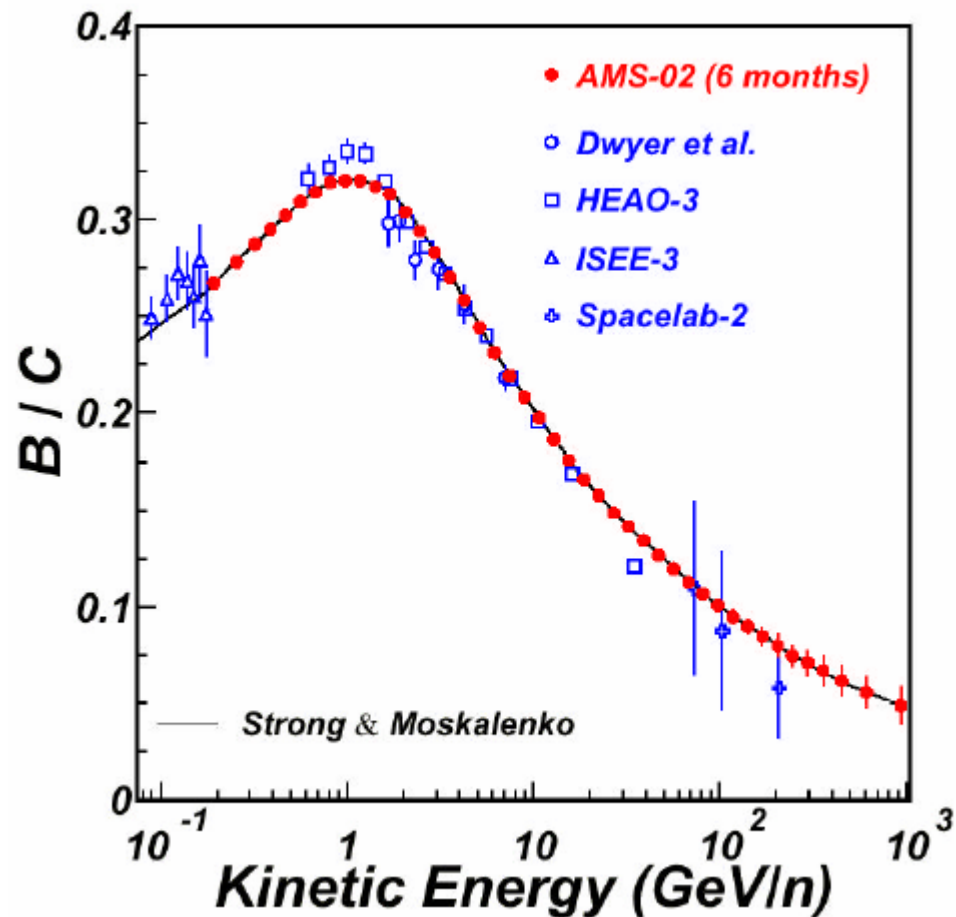


Measurement of Nuclei with AMS-02

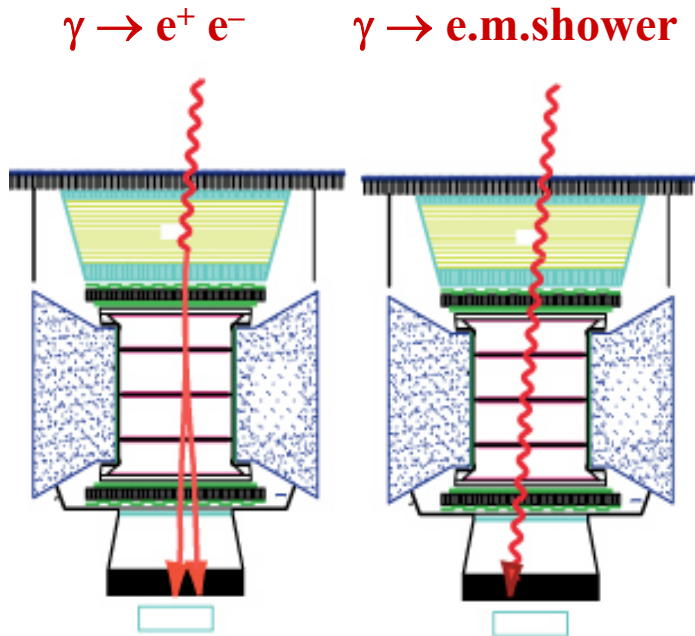
Ratio $^{10}\text{Be}/^9\text{Be}$
1 year of data taking



Ratio B/C
6 months of data taking



Unidentified Sources with AMS



AMS

- Source localization:
($E > 10 \text{ GeV}$) $< 2'$

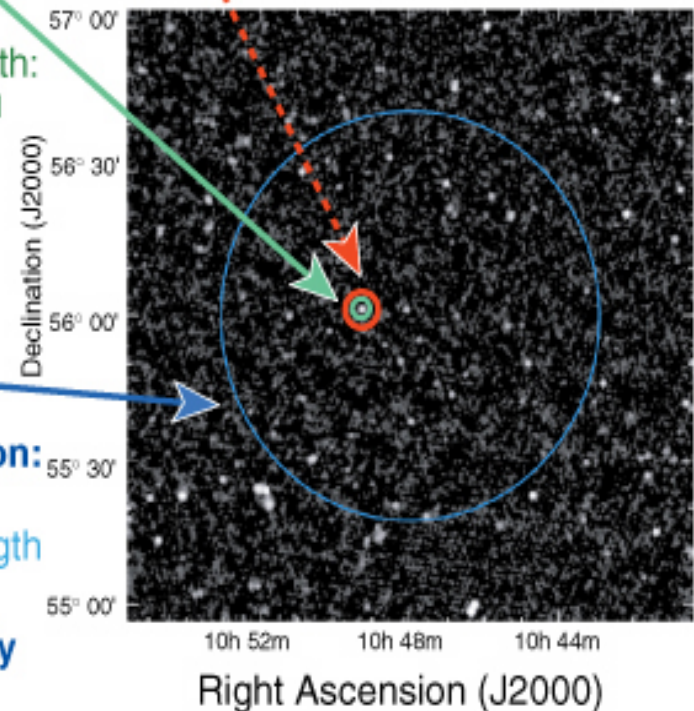
In 1 Year and
for source of strength:
 $5. \times 10^{-8} \text{ ph cm}^{-2} \text{ s}^{-1}$
($E = 1 \text{ GeV}$)

EGRET

- Source localization:
 $< 30'$
for source of strength
 $10^{-8} \text{ ph cm}^{-2} \text{ s}^{-1}$
- Limited sensitivity
above 1 GeV

GLAST

- Source localization:
 $< 5'$ and high sensitivity



AMS has a very high intrinsic resolution. Comprises a star tracker and a GPS for precise reference system

Gamma ray sources

COMPARISON OF PERFORMANCES

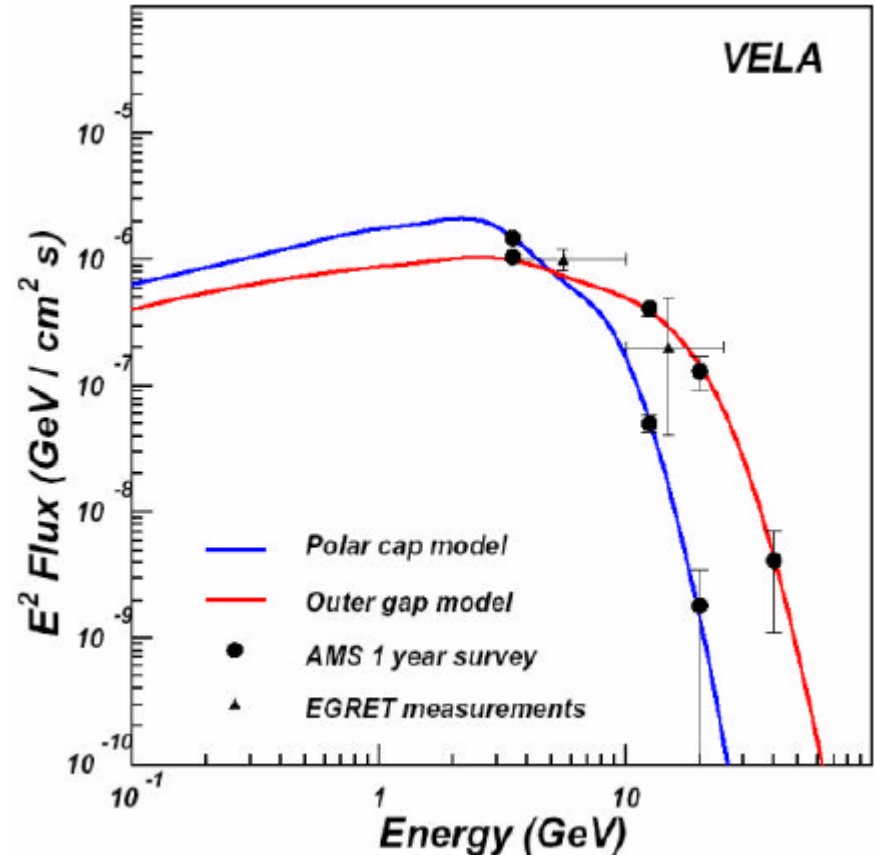
Complementarity between GLAST and AMS

Variable	EGRET	GLAST	AMS
Energy range	20MeV -	20MeV -	1GeV -
	30GeV	300GeV	1TeV
Peak effective area (cm ²)	1500	10 000	2 200
Energy Resol. (%)	10	<10	<3
Single photon Angular res.	2-3° (>1GeV)	<0.15 ° (>10GeV)	TRK : <0.8 ° (1GeV) <0.02 ° (100GeV)
F.o.V.	0.5Sr	3Sr	2Sr

EXEMPLE OF THE VELA PULSAR

1 year data taking

Allows to decide between polar cap and outer gap models



CONCLUSIONS

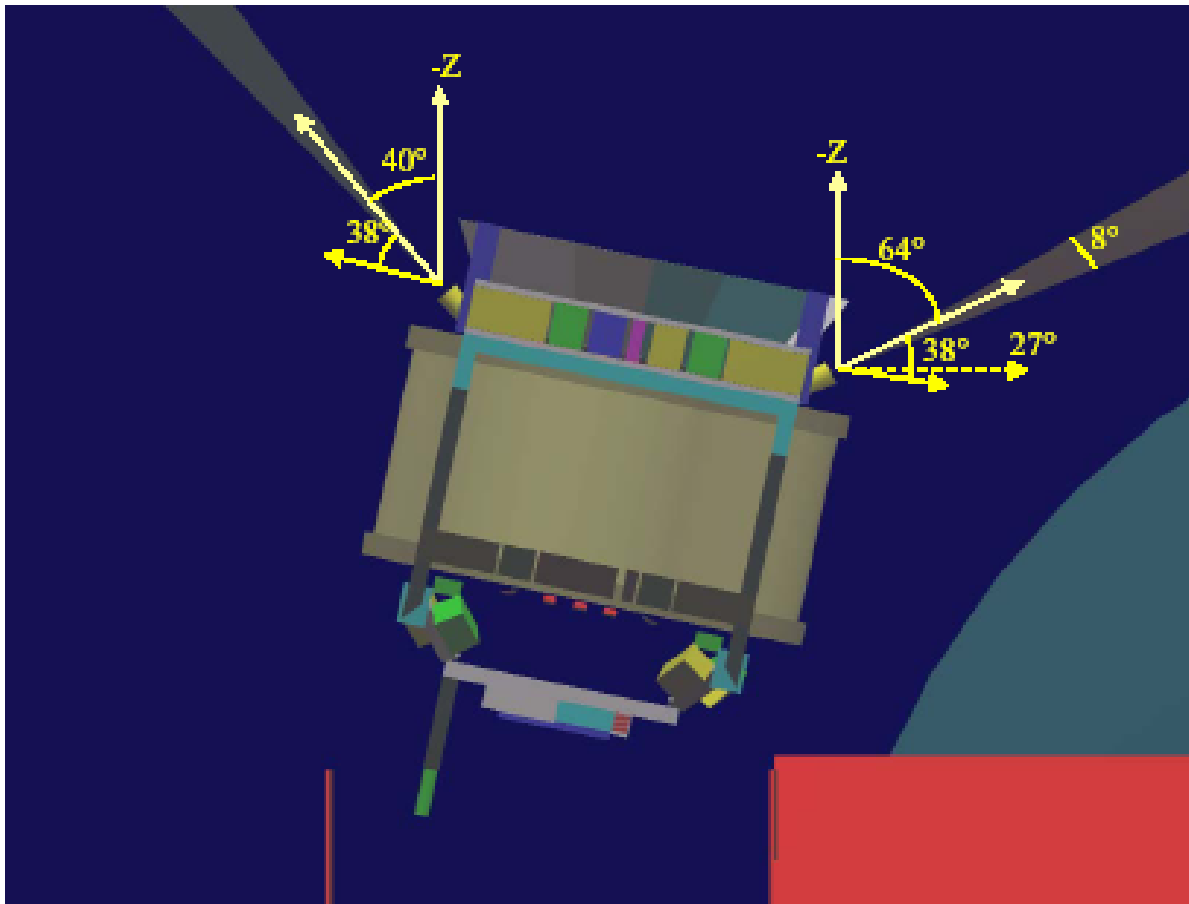
- Most of the sub-detectors will be ready by end 2004
 - Detector integration in 2005
 - Global Thermal-Vacuum test at ESA (Nordwijk, NL) end 2005 / beg. 2006
 - Then AMS is ready for launch
-
- **AMS02 will measure charged cosmic rays up to TV rigidity for 3 to 5 years on the International Space station as from 2007.**
 - To search for:**
 - Antimatter
 - Dark Matter
 - Cosmic Ray Fluxes and propagation
 - High Energy γ sources
 - **All the physics channels are measured in the same conditions and simultaneously, which will give a strong constraint on models and increase the potential of discovery.**

AMS-02 on ISS (ARTIST VIEW)



v03K092Ken05

Star Tracker



Angular
resolution

30 arcsec

2 CCD Camera
at 90°